

Additive Manufacturing of Metals and Alloys

10. Properties of lattice structures

February 2023

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Properties of lattice structures

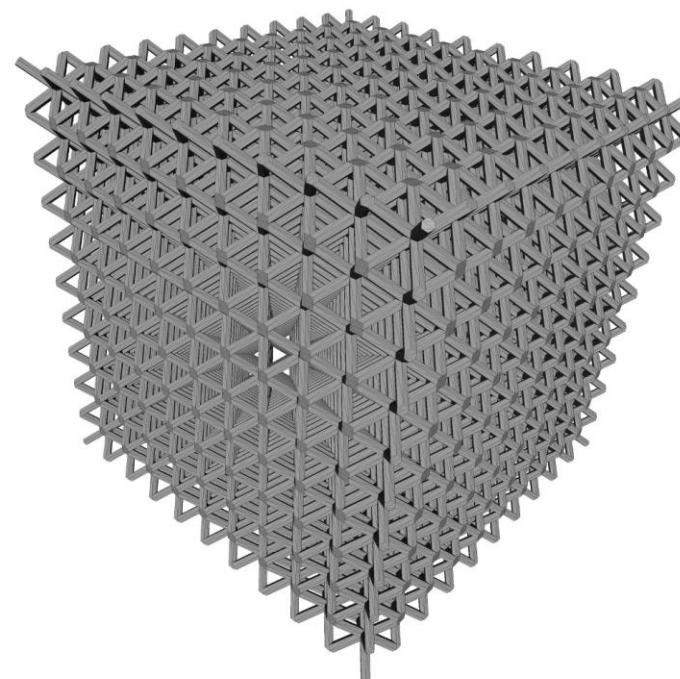
- What are lattice structures?
 - Definition
 - Multiscale and multifunctional materials
- Classification of lattice structures
 - bending dominated
 - stretching dominated
- AM of lattice structures: ongoing issues
 - surface roughness
 - effect of build orientation
 - post-processing
- Mechanical properties of AM lattice structures
 - static
 - dynamic
 - effect of post-processing
 - effect of loading conditions

Cellular structures include lattices and foams

Cellular structures are defined as a combination of material and space. A cellular material is a connected 3D network of hollow or dense struts.



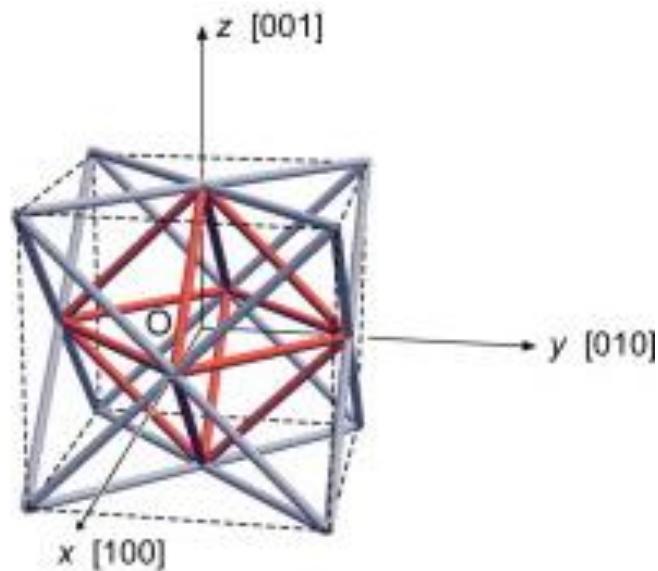
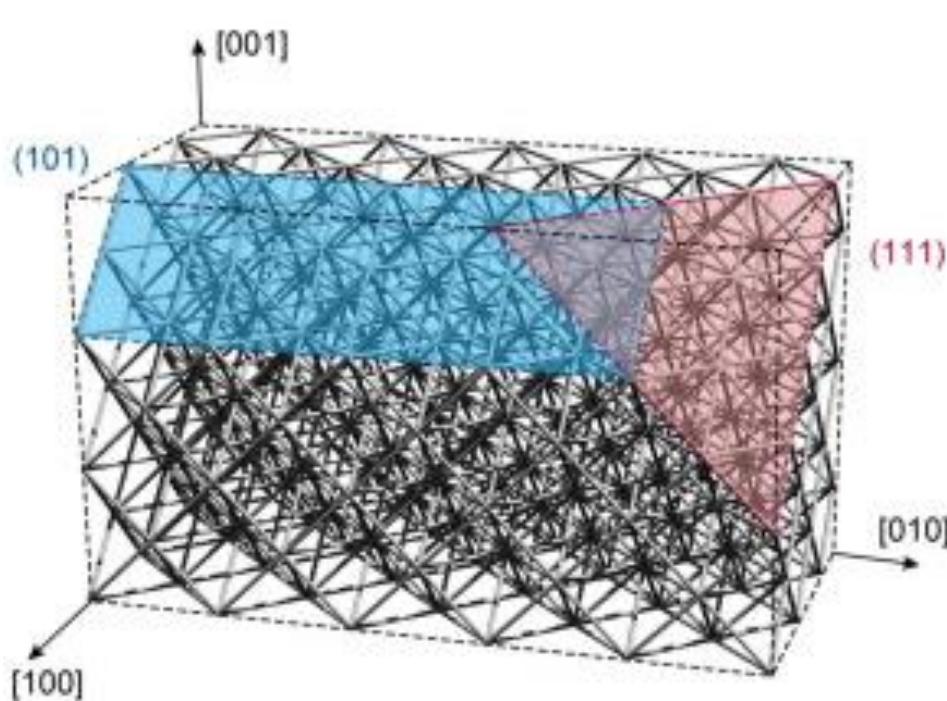
A **foam** is a **stochastic** cellular material



A **lattice structure** is a **periodic** arrangement of repeating unit cells

What are lattice structures?

Lattice structures are formed by the three-dimensional repetition of a **unit cell**.



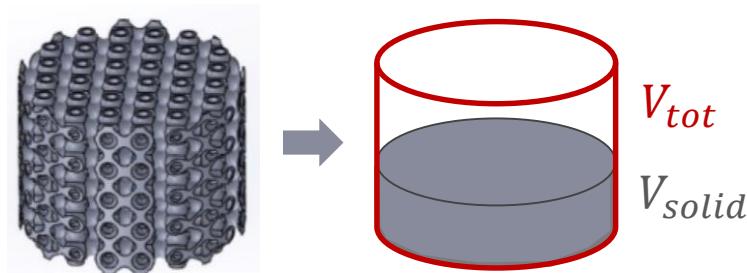
An octet-truss lattice structure and its unit cell

What are lattice structures?

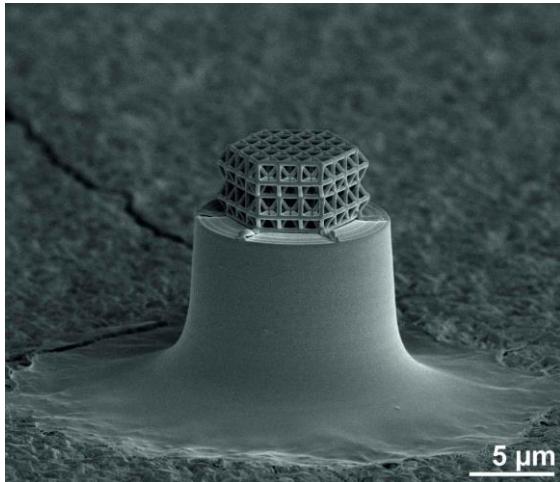
Properties of lattice structures depend on:

- (i) the material of which the structure is made,
- (ii) the shape and connectivity of the struts (Maxwell's criterion),
- (iii) the relative density of the structure $\rho_{rel} = \frac{\tilde{\rho}}{\rho_s}$
where $\tilde{\rho}$ is the density of the structure
and ρ_s that of the solid of which it is made.

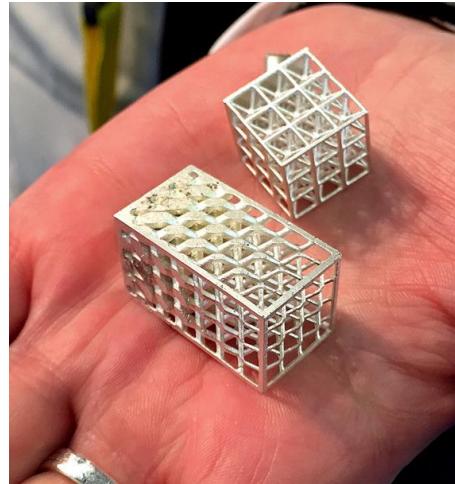
$$\rho_{rel} = \frac{\tilde{\rho}}{\rho_s} = \frac{V_{solid}}{V_{tot}}$$



Lattice structures are multiscale materials



carbon nanolattice



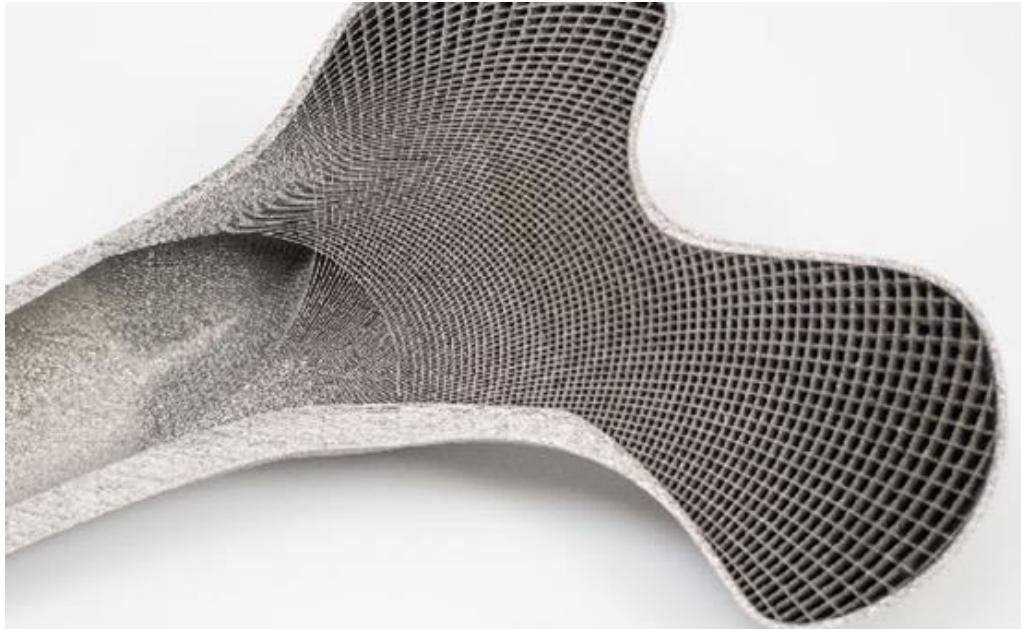
metal lattice



orthogonal timber lattice



Lattice structures are multifunctional materials



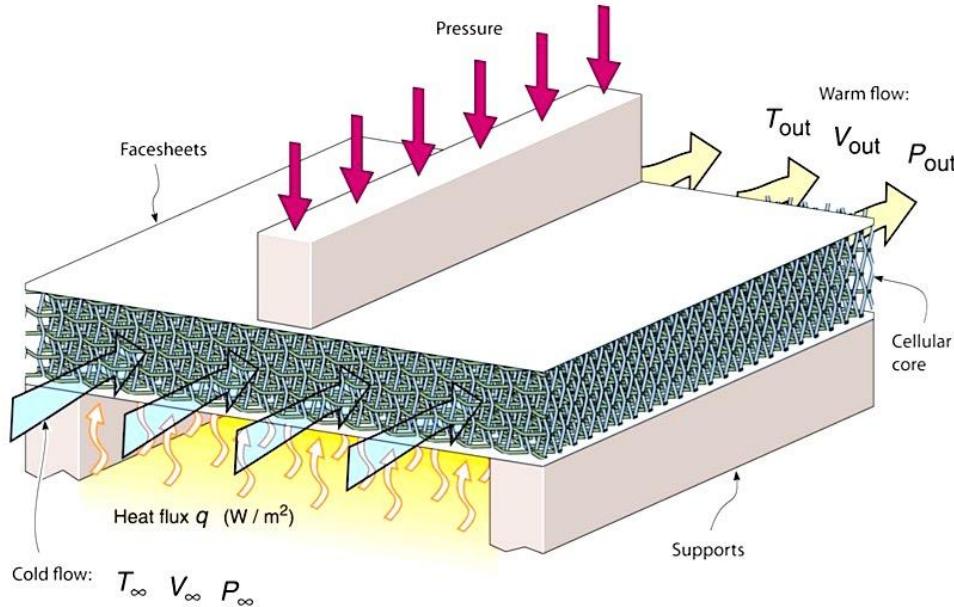
Integration of a porous structure in an implant:
⇒ stiffness reduction,
⇒ no stress shielding



Interconnected porosity:
⇒ favoured vascularization
and bone growth

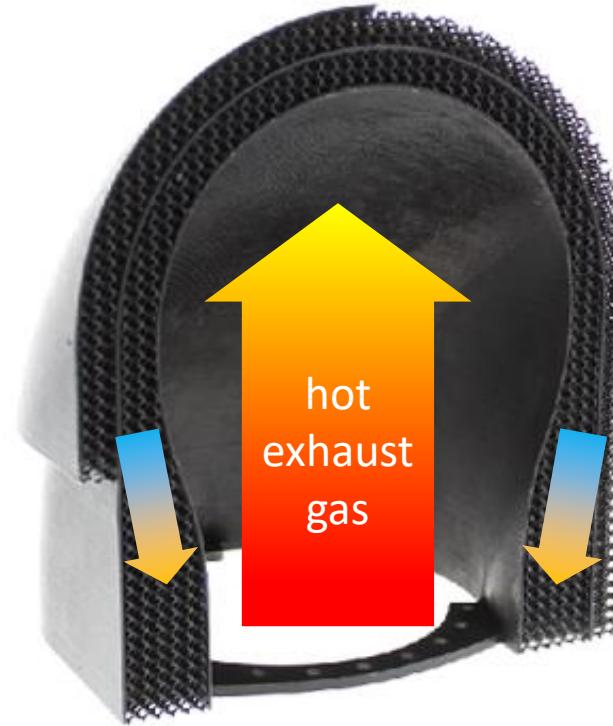
Biocompatibility and long-term osseointegration

Lattice structures are multifunctional materials



Heat exchanger using a cellular core sandwich panel

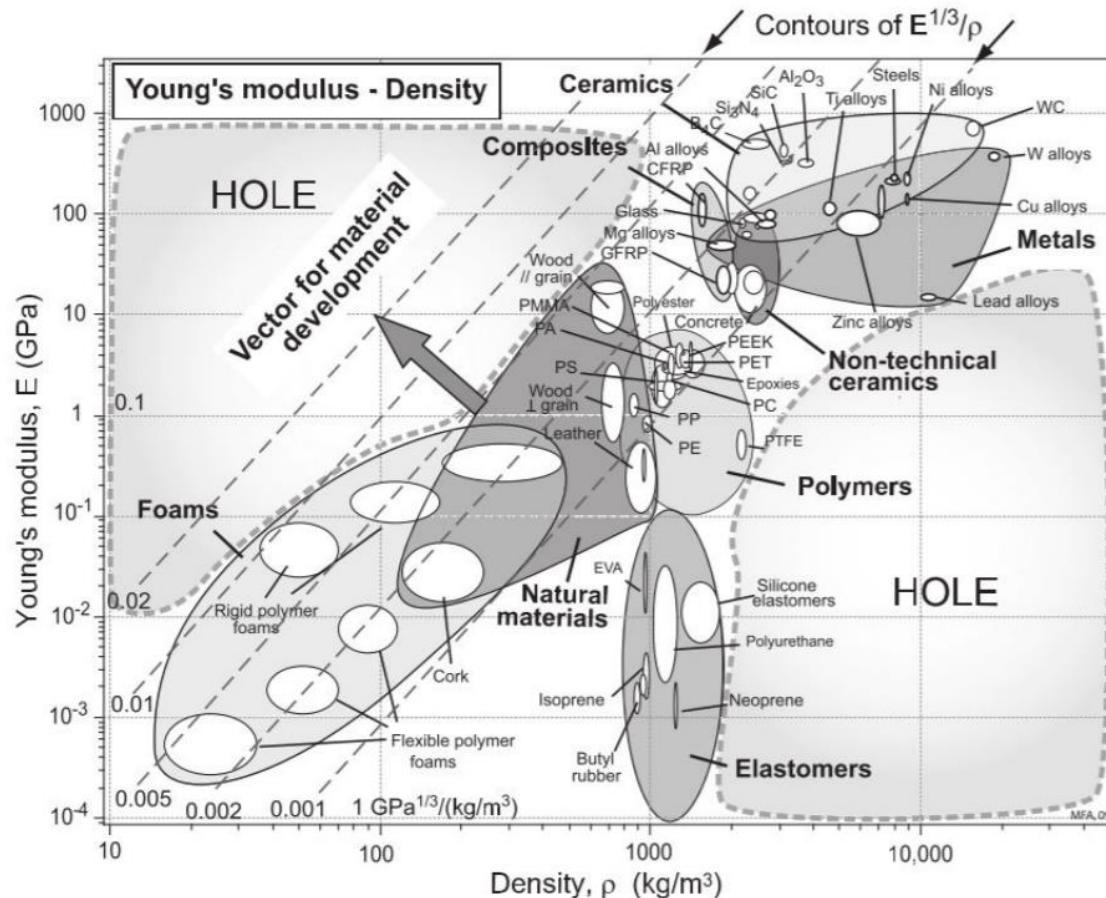
Very high surface area
⇒ heat transfer to the coolant



Inconel 625 prototype helicopter exhaust nozzle.

Lattice structures allow engineers to « fill holes in the material-property space ».

Example: in order to produce parts with **higher stiffness-to-density ratio**, monolithic materials are not sufficient and the development of architected materials is needed.



Properties of lattice structures

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- **Classification of lattice structures**
 - bending dominated
 - stretching dominated
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 - effect of build orientation
 - post-processing
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Classification of lattice structures

Two types of lattice structures can be distinguished:

- bending-dominated
- stretching-dominated.

This distinction is based on the **connectivity** of the structure.

Maxwell's stability criterion :

$$M = b - 2j + 3, \text{ in 2D}$$

$$M = b - 3j + 6, \text{ in 3D}$$

b: number of struts

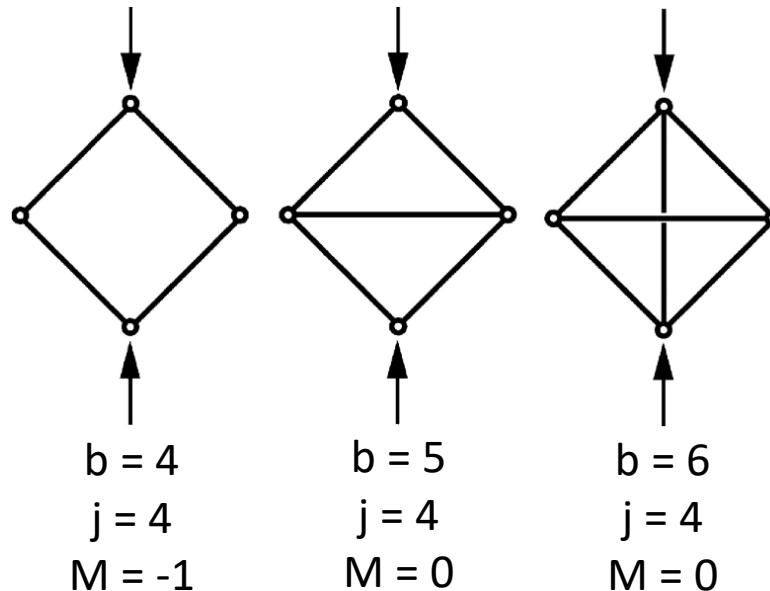
j: number of nodes

in a unit cell

M < 0 : bending dominated

M = 0 : stretching dominated

M > 0 : over-constrained



Classification of lattice structures

Maxwell's stability criterion :

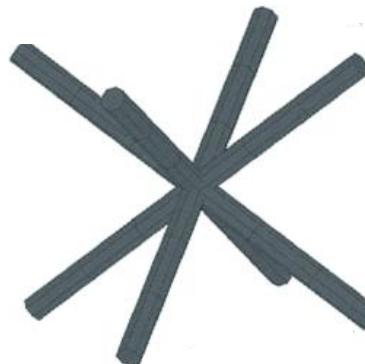
$$M = b - 2j + 3, \text{ in 2D}$$

$$M = b - 3j + 6, \text{ in 3D}$$

b: number of struts

j: number of nodes

BCC



$$b = 8$$

$$j = 9$$

$$M = -13$$

BCC with Z strut

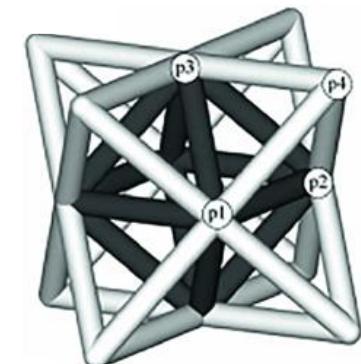


$$b = 12$$

$$j = 9$$

$$M = -9$$

Octet-truss



$$b = 36$$

$$j = 14$$

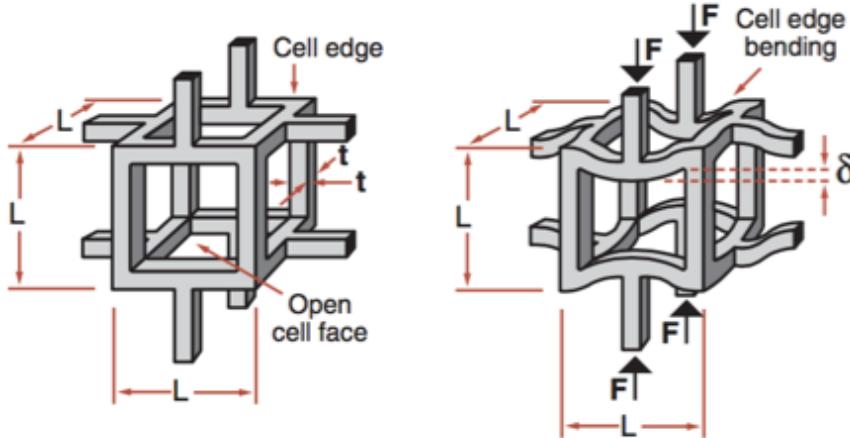
$$M = 0$$

$M < 0$: bending dominated

$M = 0$: stretching dominated

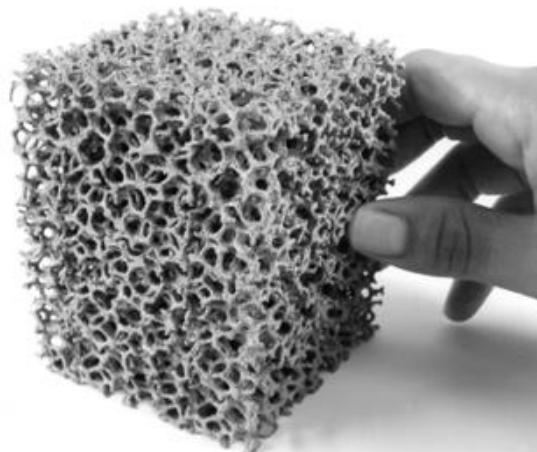
$M > 0$: over-constrained

Bending-dominated structures



If $b - 3j + 6 < 0$, the **low connectivity** of the struts allows them to bend when the lattice is loaded.

Foams are typical examples of bending-dominated cellular structures.

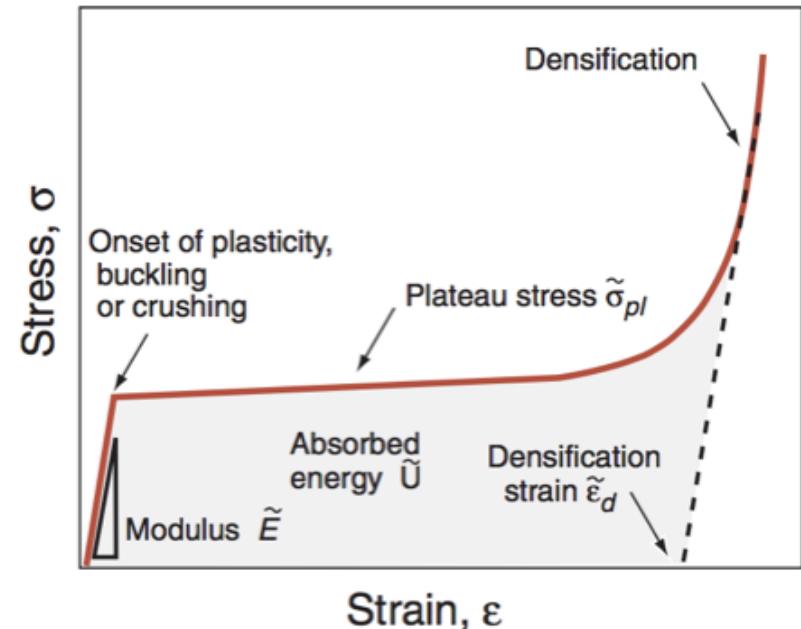


Bending-dominated structures

Estimation of the useful **energy** that a foam can absorb per unit volume:

$$\tilde{U} \approx \tilde{\sigma}_{pl} \tilde{\varepsilon}_d,$$

where $\tilde{\sigma}_{pl}$ is the plateau stress, and $\tilde{\varepsilon}_d$ the densification strain.



Stretching-dominated structures

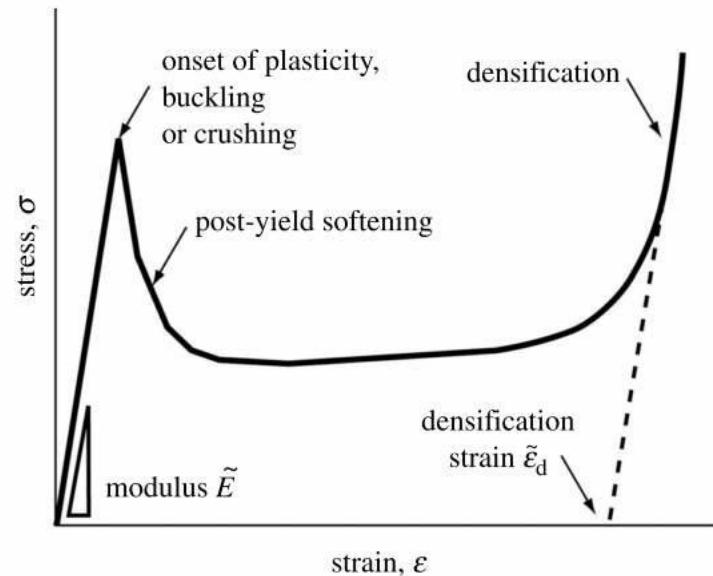
If $b - 3j + 6 \geq 0$, the high nodal connectivity of the structure suppresses bending, forcing the struts to stretch.

Isotropic behaviour: one third of the struts carries tension when it is loaded in simple tension

$$\frac{\tilde{E}}{E_s} \approx \frac{1}{3} \left(\frac{\tilde{\rho}}{\rho_s} \right) \quad \text{where } \tilde{E} \text{ is the stiffness of the structure}$$

and E_s the Young's modulus of the solid of which it is made

Quasi-static compression behaviour:



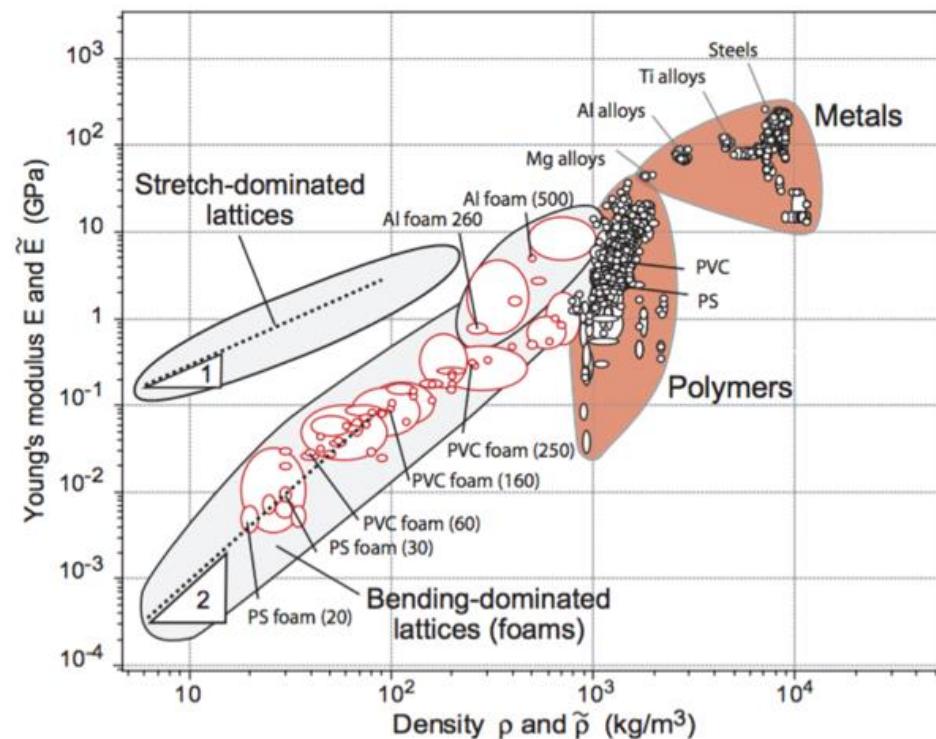
Bending vs stretching-dominated structures

Stretch-dominated lattices: $\frac{\tilde{E}}{E_s}$ scales as $\frac{1}{3} \frac{\tilde{\rho}}{\rho_s}$

⇒ weight efficient, **stress-supporting** structures

Bending-dominated lattices: $\frac{\tilde{E}}{E_s}$ scales as $\left(\frac{\tilde{\rho}}{\rho_s}\right)^2$

⇒ more compliant and **weaker** structures

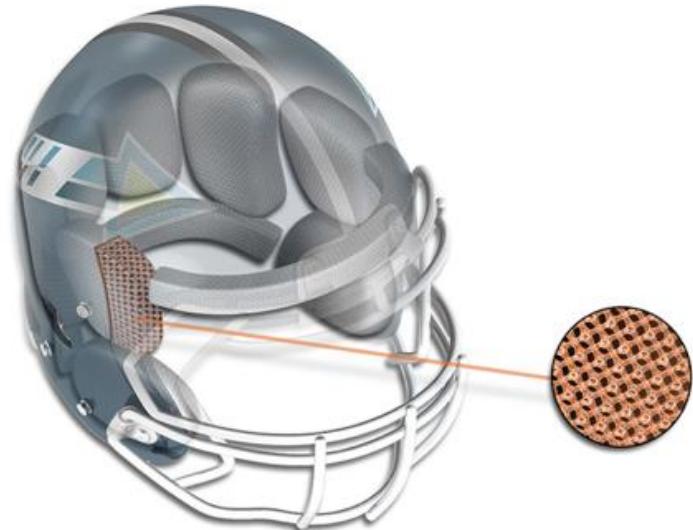
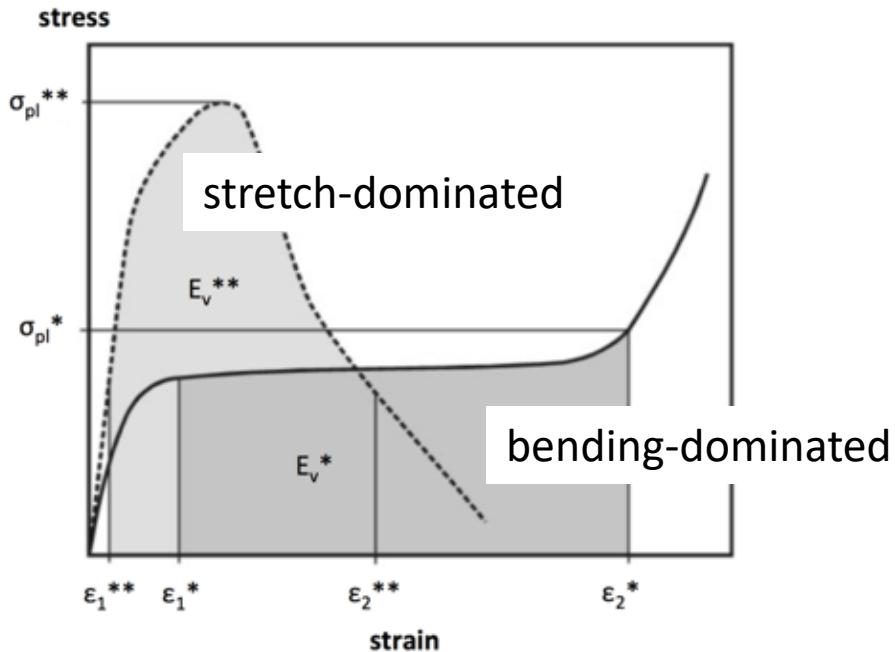


Applications of stretching-dominated structures



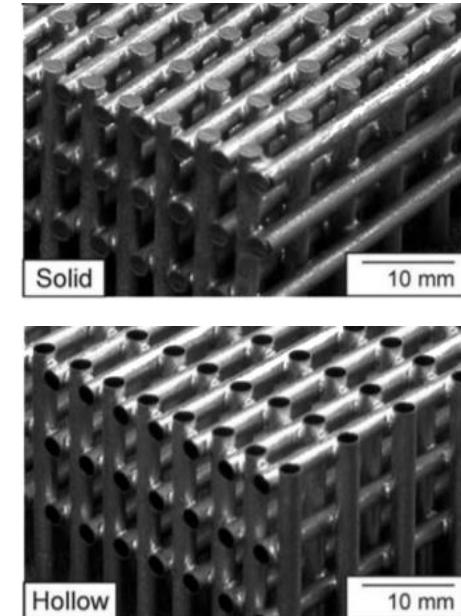
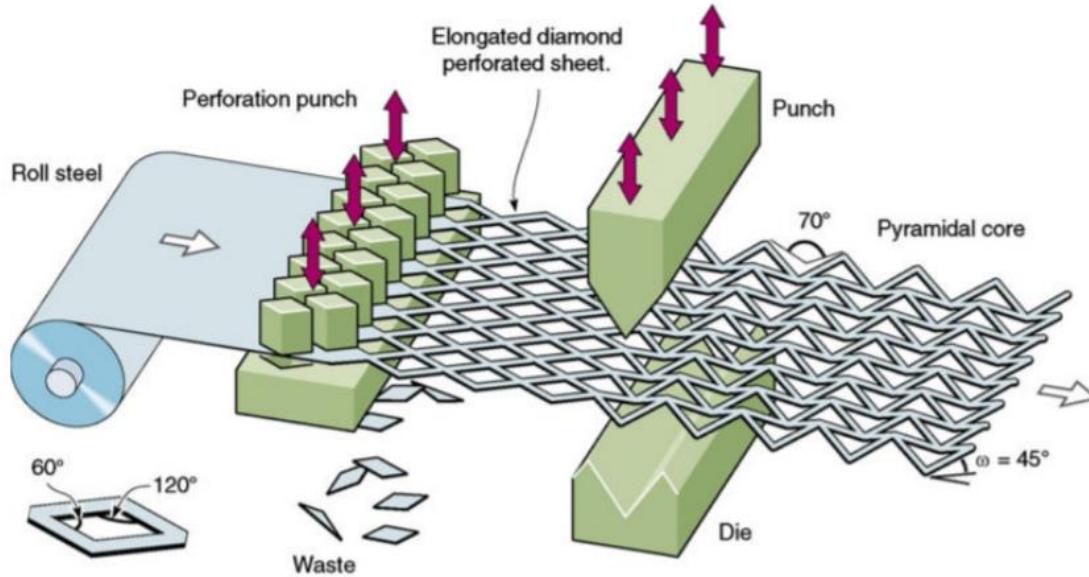
High strength-to-weight and stiffness-to-weight ratios
⇒ substantial **mass decrease** by incorporating these structures
where the maximum stresses are low in **aerospace** components

Applications of bending-dominated structures



⇒ energy absorption applications: packaging, protection against impact...

The manufacturing of lattice structures by conventional techniques is **complex**



Production from an expanded metal sheet

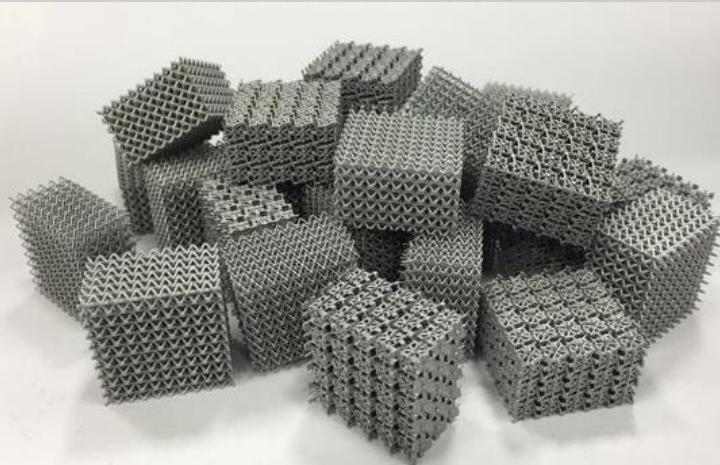
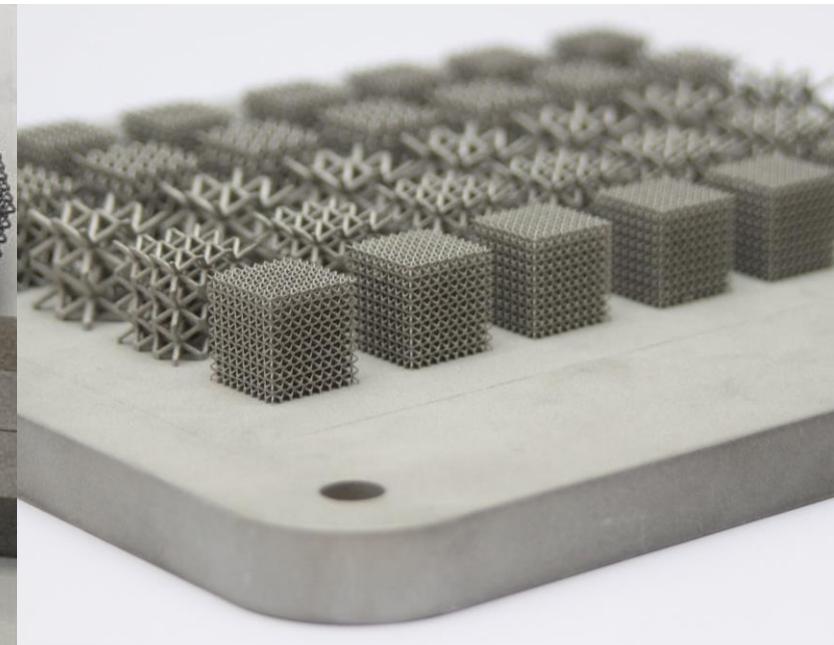
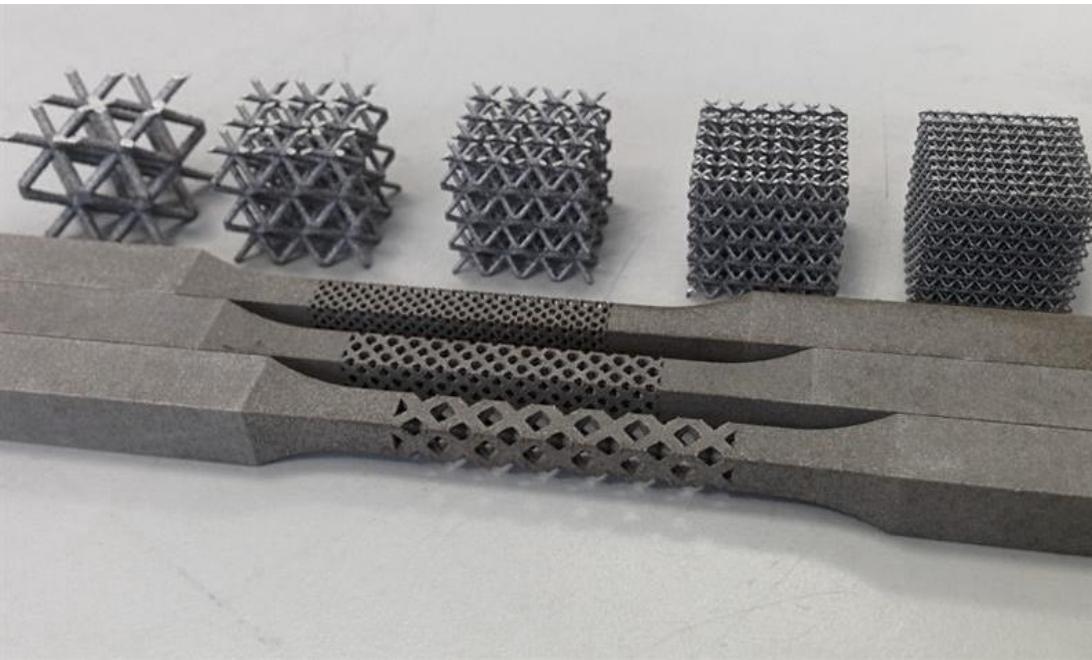
- 1) perforation
- 2) folding
- 3) stacking

Produces a **large amount of waste** and requires the use of an alloy that is **ductile** enough to be perforated and punched.

Solid or hollow metallic wire assembly

Permits only the production of a **limited amount of designs**.

AM is particularly well suited for the production of metallic lattice structures

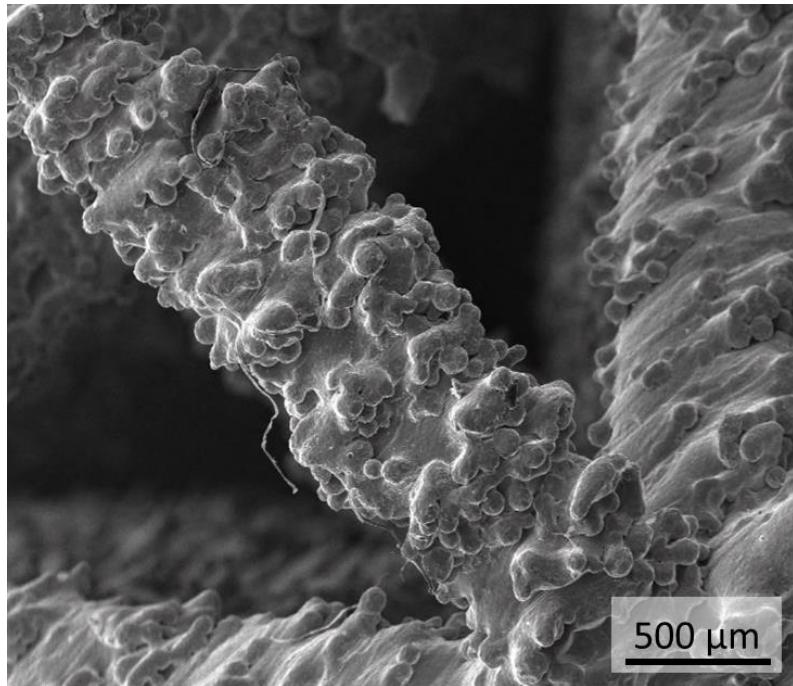


Properties of lattice structures

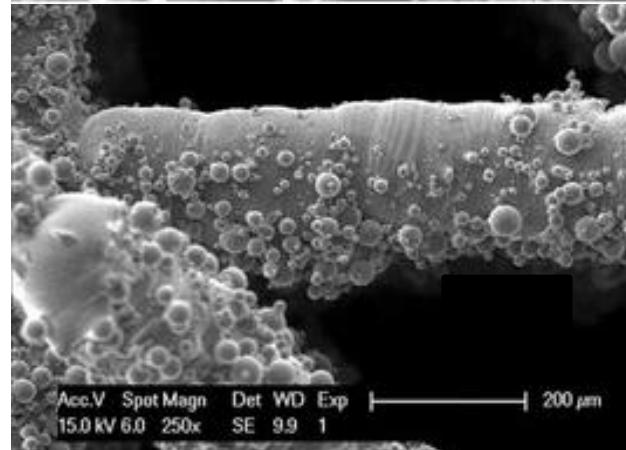
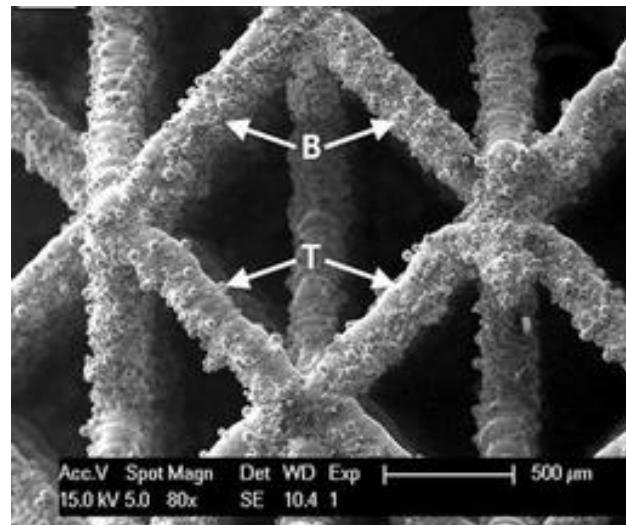
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Surface roughness: EBM vs LPBF

EBM

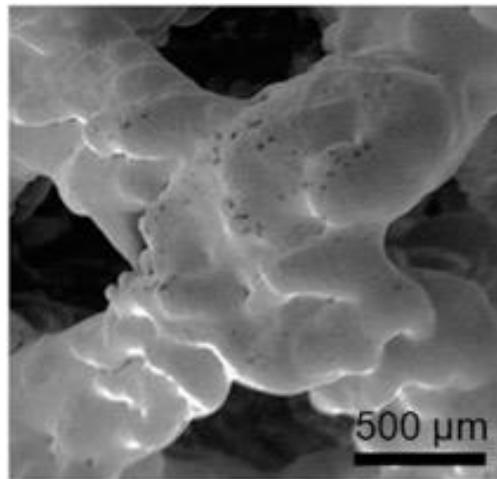


LPBF

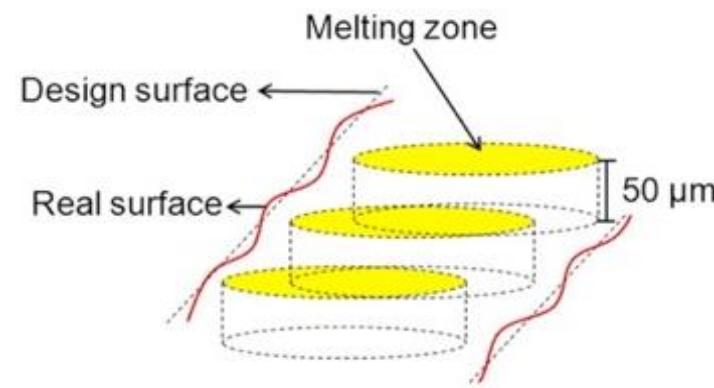
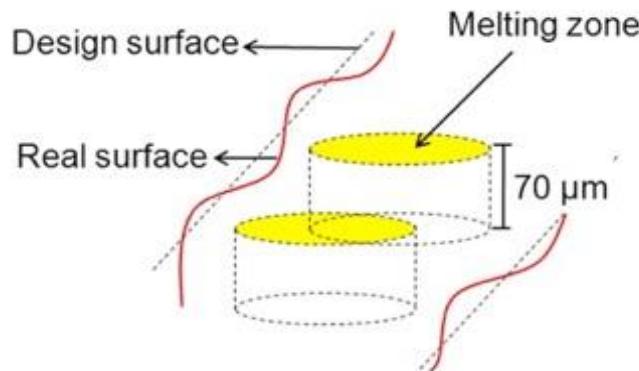
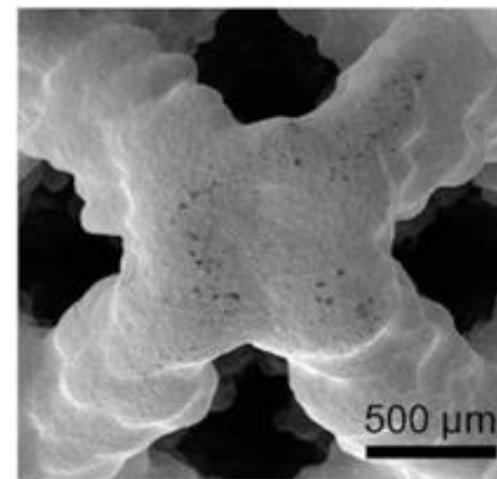


Surface roughness: staircase effect

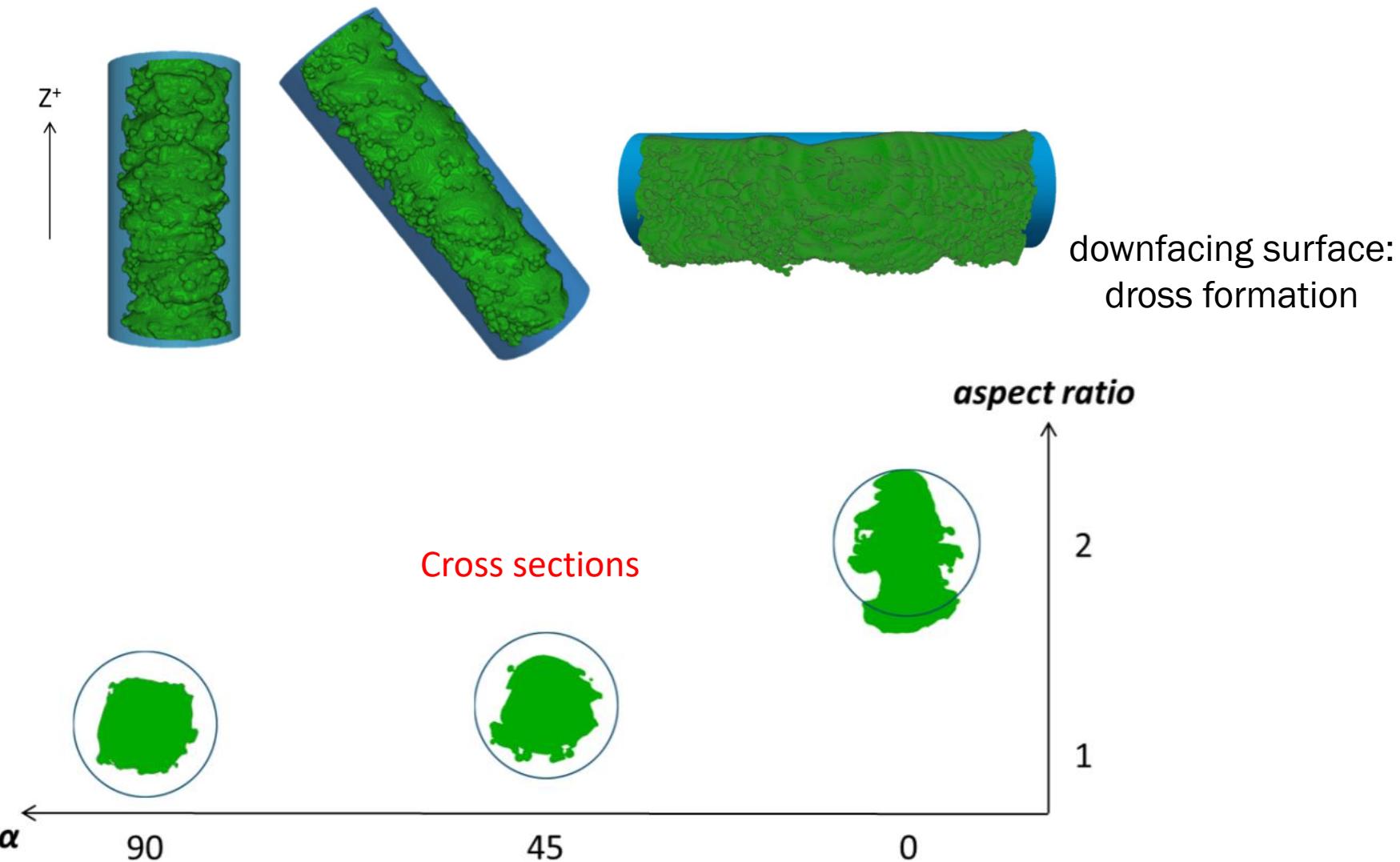
EBM



LPBF

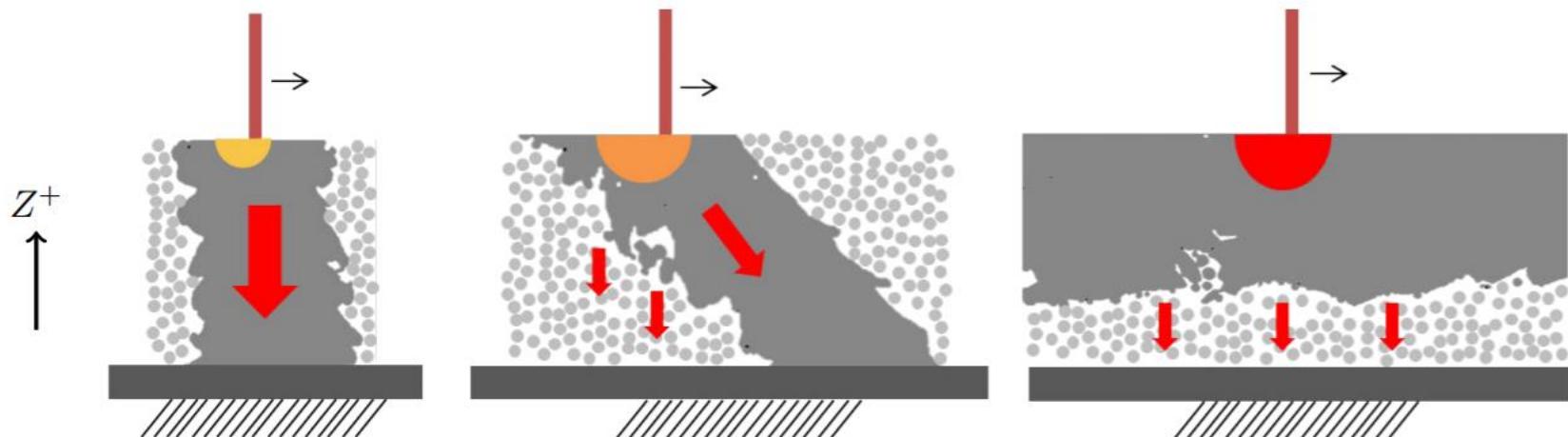


Surface roughness: effect of strut inclination



Surface roughness: effect of strut inclination

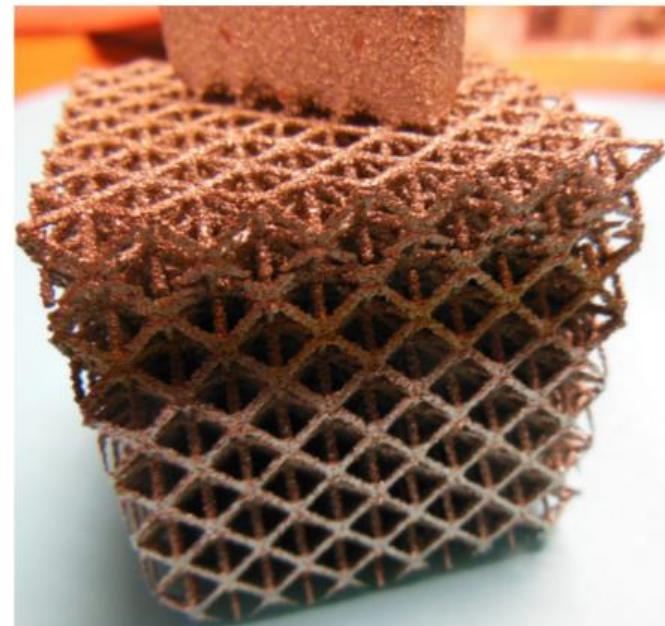
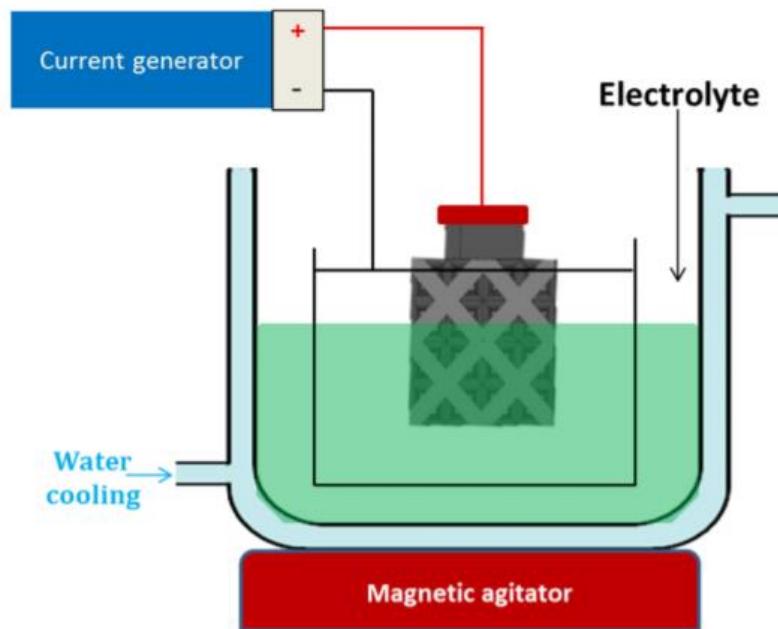
Thermal behavior during processing for different orientations:



Electro-chemical polishing ECP

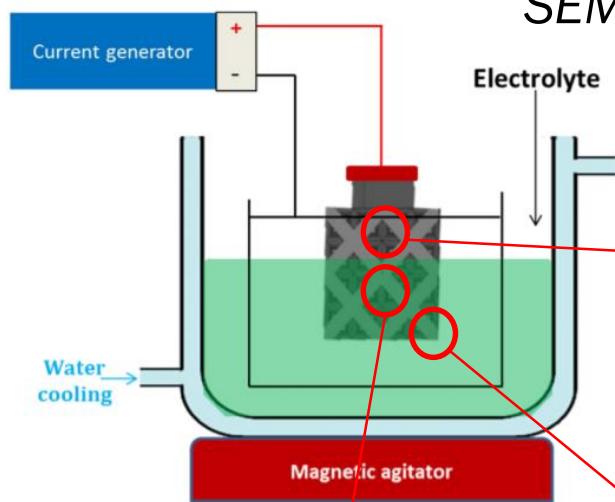
Ti-6Al-4V

Experimental set-up

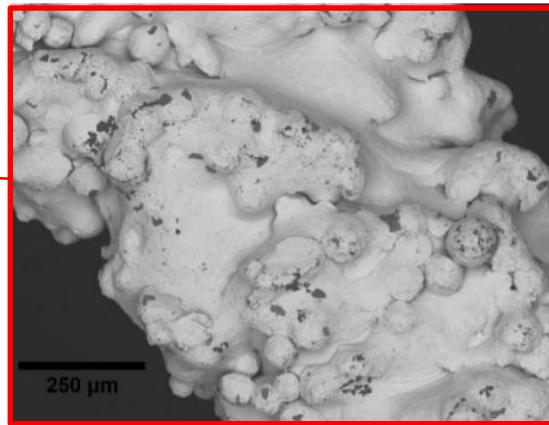


Electrolyte: vol10% of perchloric acid and vol90% of acetic acid

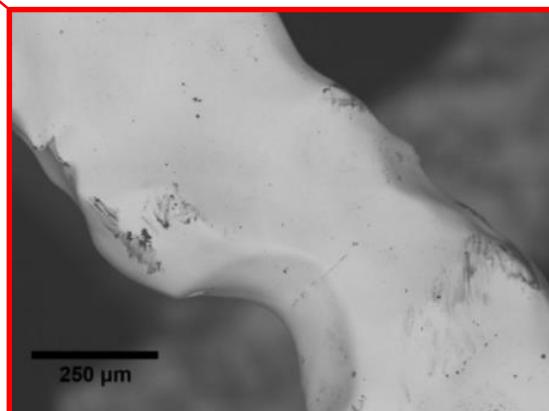
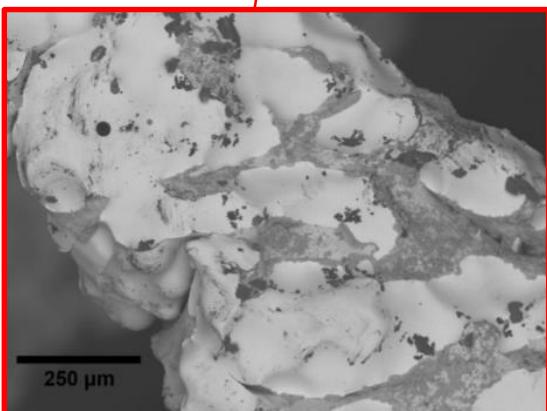
Electro-chemical polishing ECP does not allow to homogeneously reduce surface roughness



SEM micrographs of external struts after ECP

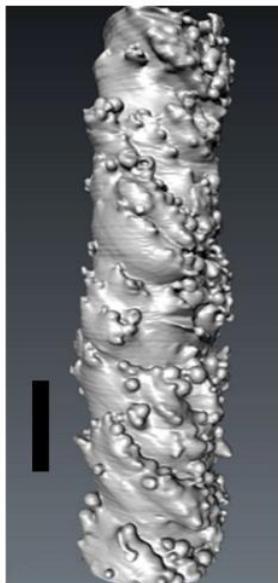


Raw strut in the emerged zone

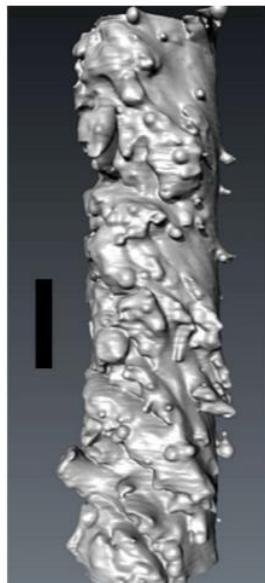


Electro-chemical polishing ECP does not allow to homogeneously reduce surface roughness

X-ray tomography on the struts in different zones of the lattice structure.



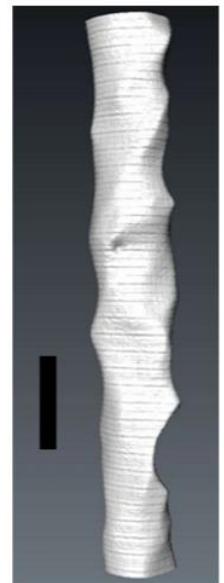
raw



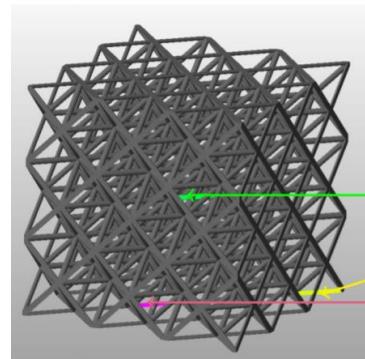
internal



bottom



bottom corner



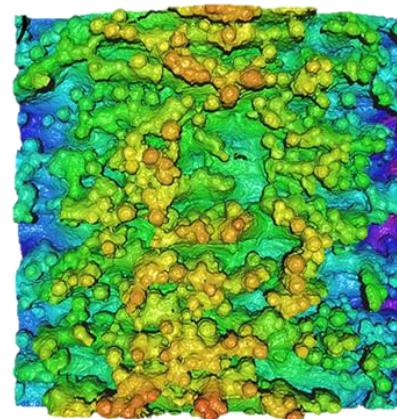
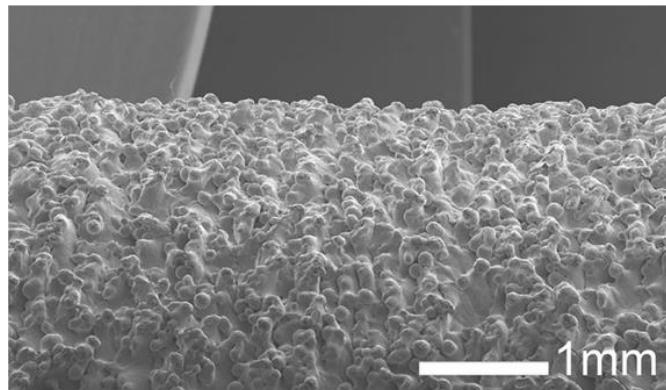
scale bar: 500 μm

Large **inhomogeneity of polishing**:

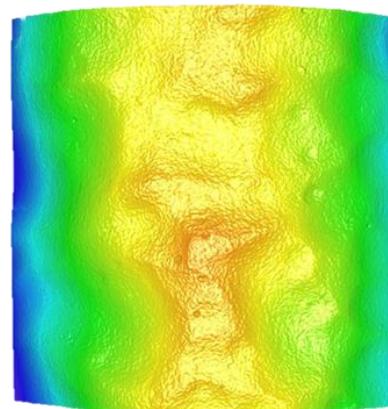
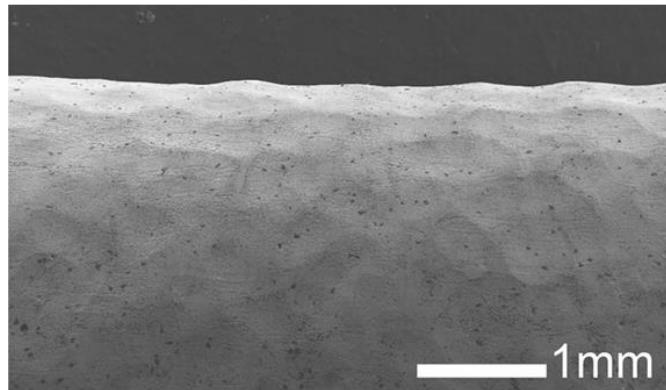
more significant roughness reduction **at the edges** of the lattice

Chemical etching

Chemical etchant: 3%HF, 13% HNO_3

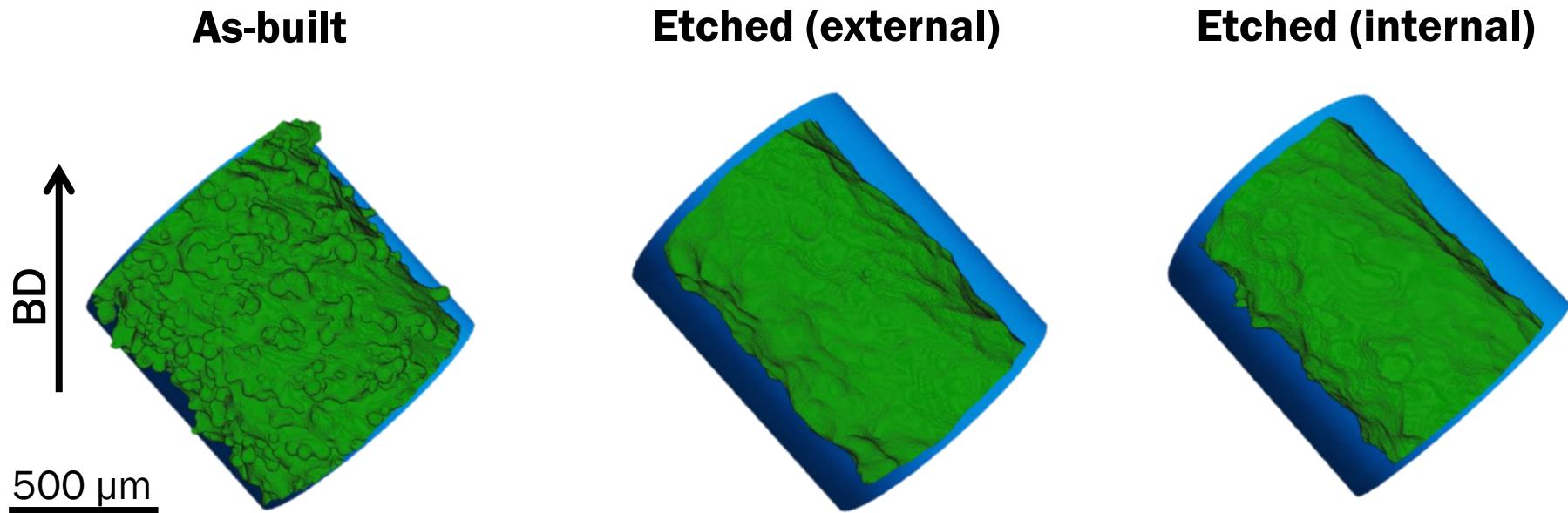


↓
chemical
etching



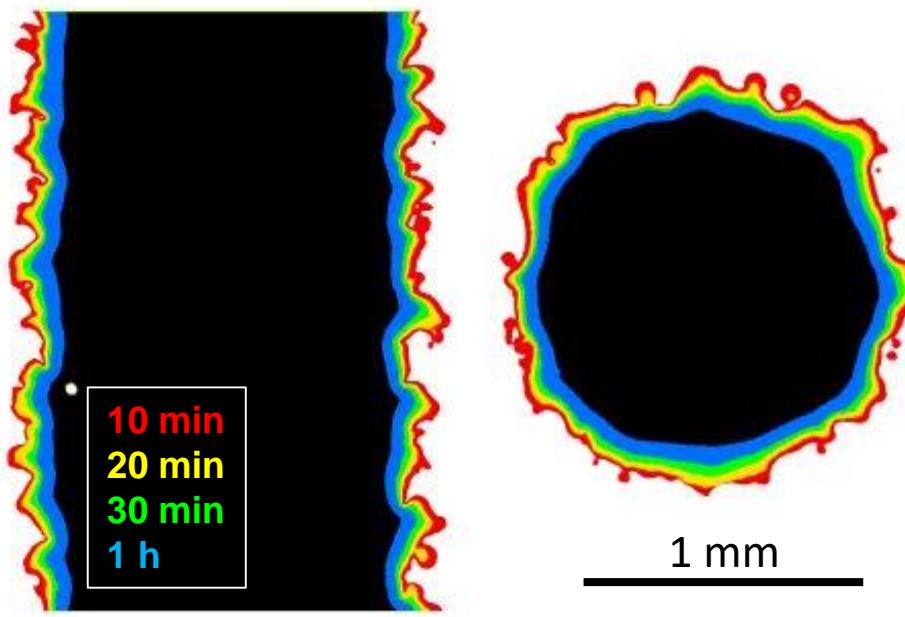
Chemical etching

Chemical etchant: 3%HF, 13% HNO_3



The chemical etching is **homogeneous throughout the structure.**

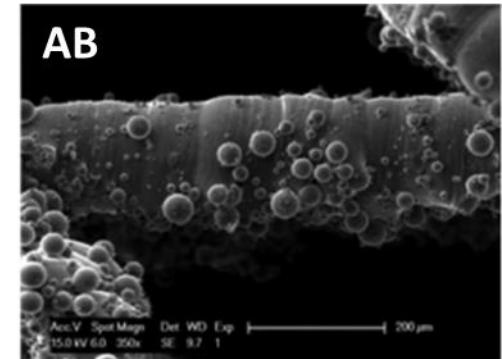
Chemical etching: effect of etching time



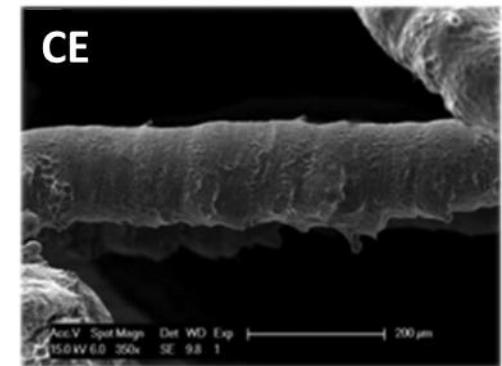
The grains of powder stuck to the surface are the first to be removed.

Combination of chemical etching and ECP

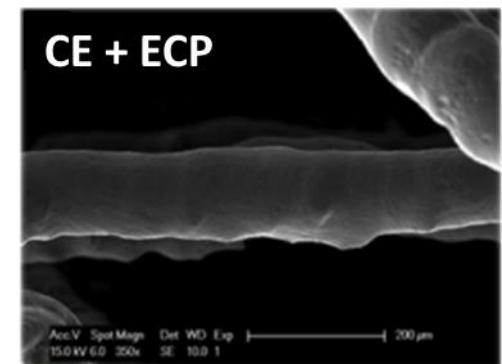
The combination of etching and electrochemical polishing with HF-based solutions has proven to be effective to obtain a **more controlled and homogeneous roughness** throughout the structures .



Chemical etching is mainly used to **remove** the partially melted or partially sintered **particles** on the surface.



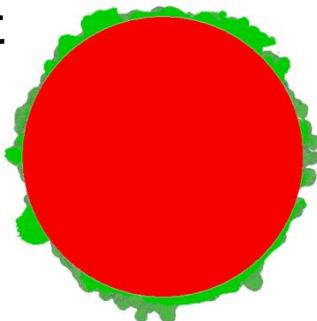
Then, **electrochemical polishing** is used to make the **surface** of the part more uniform and **smooth**.



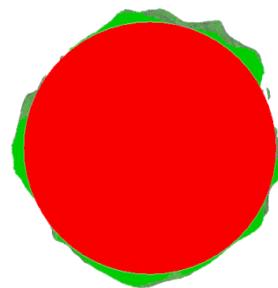
CE + ECP leads to a significant **loss in strut thickness**, which should be taken into account anticipatively in the design of the structure.

Etching results in a mechanical behavior closer to the « ideal » FEM predictions

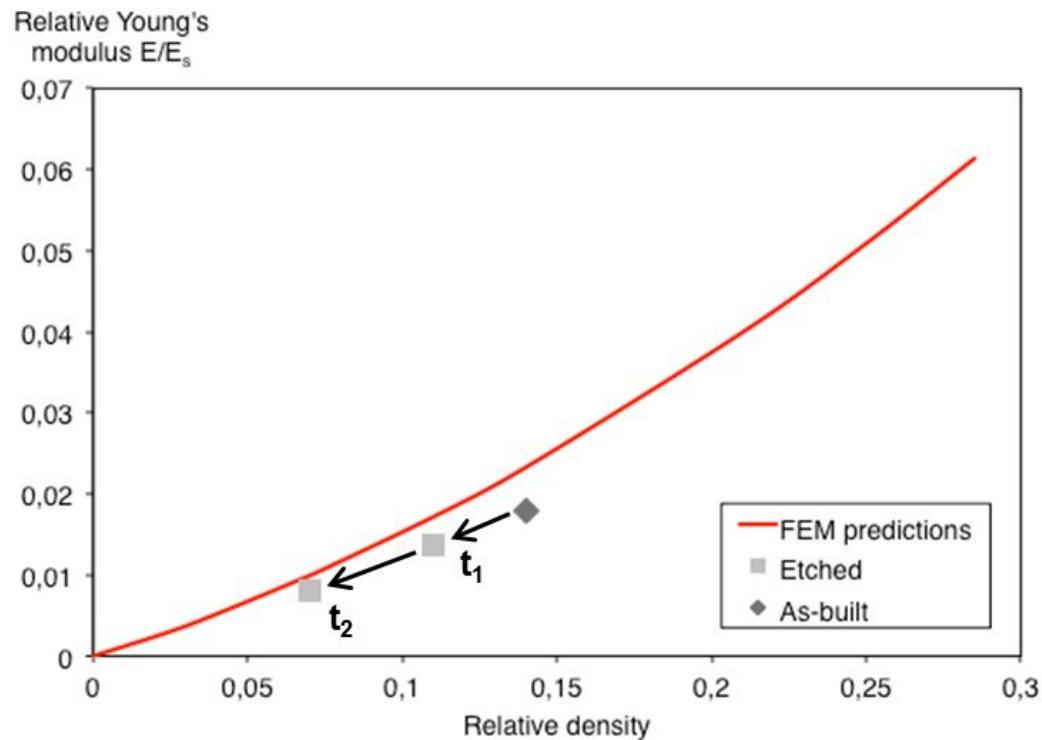
As-built



Etched



500 μ m

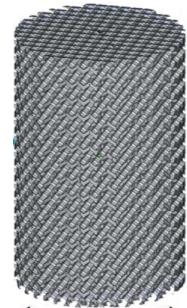


Chemical etching results in an increase in the proportion of mechanically efficient matter

The relative difference between the measured and the **theoretical stiffness** decreases as the etching time increases.

Effect of post-processing on fatigue

Experimental approach



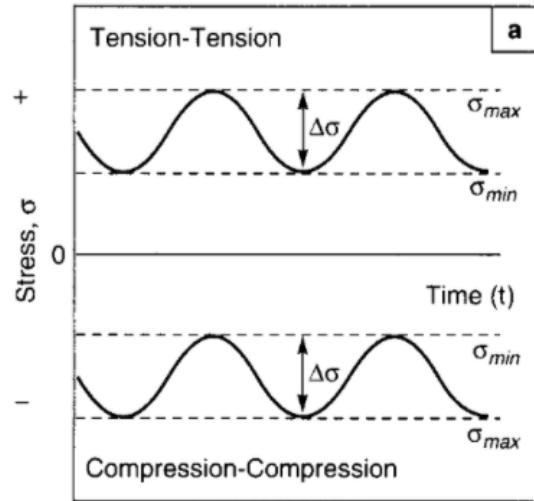
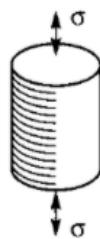
**AS-BUILT
SCAFFOLD**

Ultrasonic cleaning
Thermal treatment: SR, HIP
Surface treatment: CE, ECP

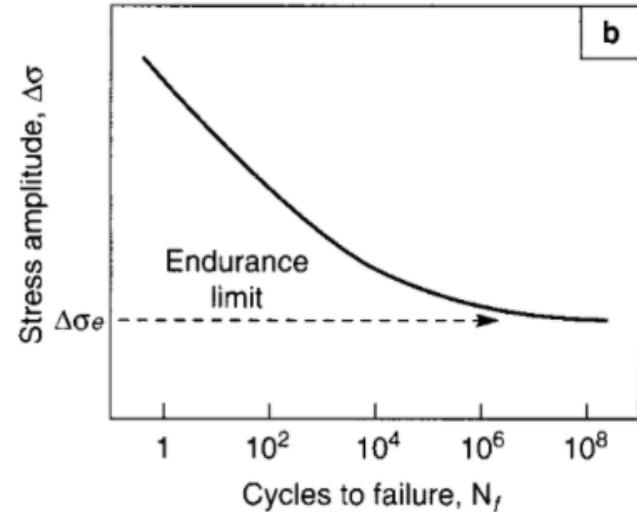


**FINAL
SCAFFOLD** $\rho_{rel_f} < \rho_{rel_AB}$

Fatigue testing

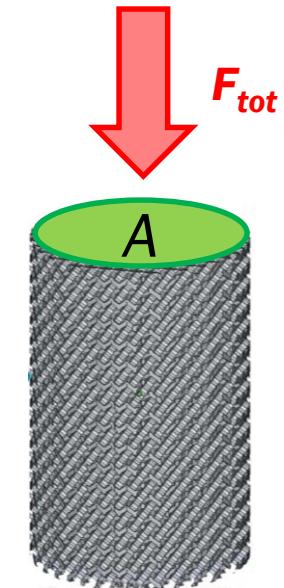
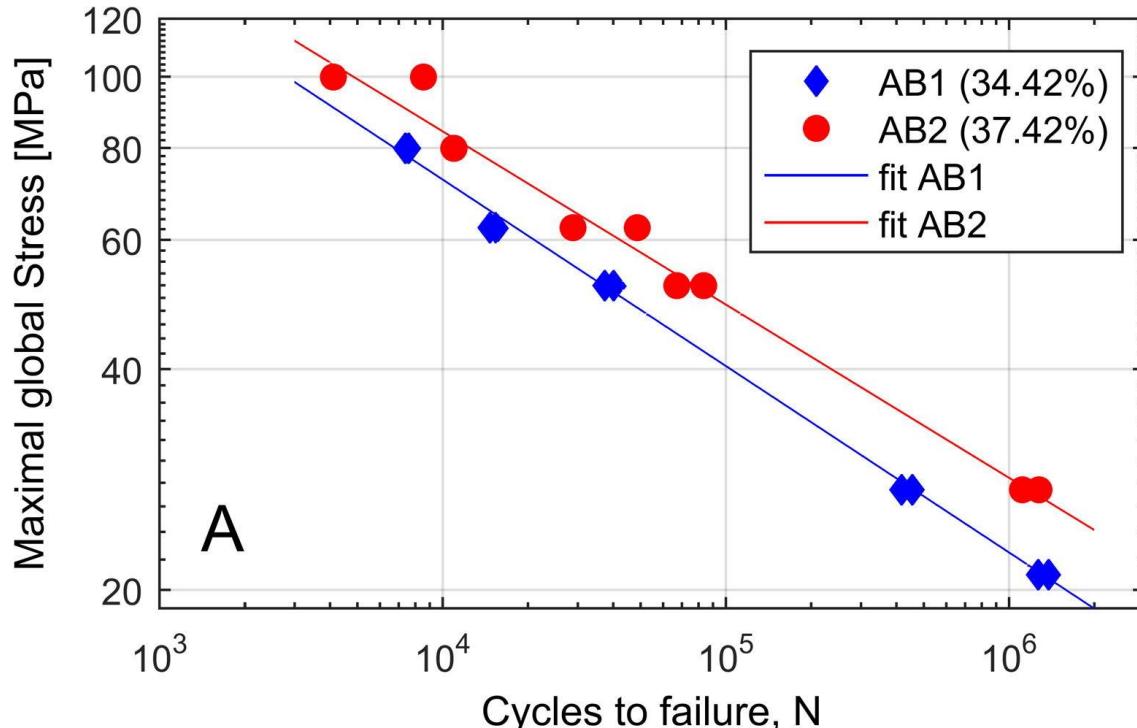


Wöhler curve



Global stress approach

Behavior of the porous material as a whole: a higher relative density results in an increased fatigue life.



$$\sigma = \frac{F_{tot}}{A}$$

How do we account for the difference in relative density when comparing the **fatigue behavior before and after post-processing**?

Local stress method

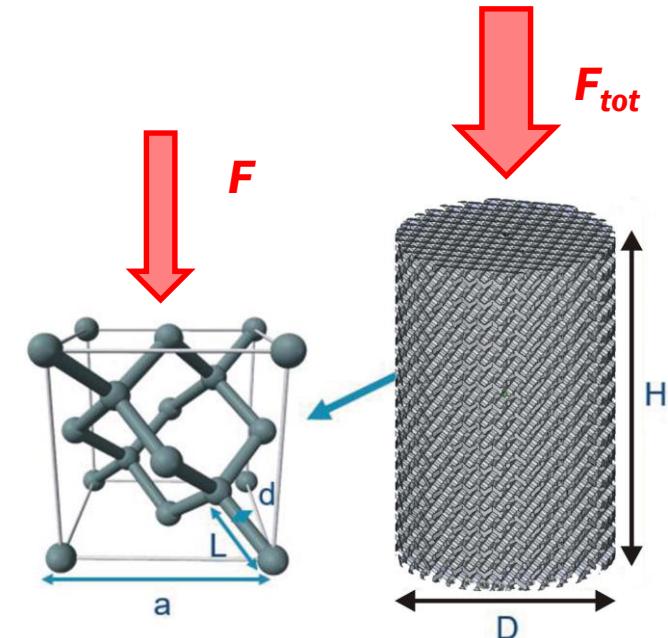
$$S_B = \frac{My}{I} = \frac{FLd \cos \theta}{4I} \quad \text{with:}$$

$$I = \frac{\pi d^4}{64} \quad y = \frac{d}{2} \quad M = \frac{1}{2} FL \cos \theta$$

$$S_A = \frac{F \sin \theta}{A} \quad \text{with} \quad A = \frac{\pi d^2}{4}$$

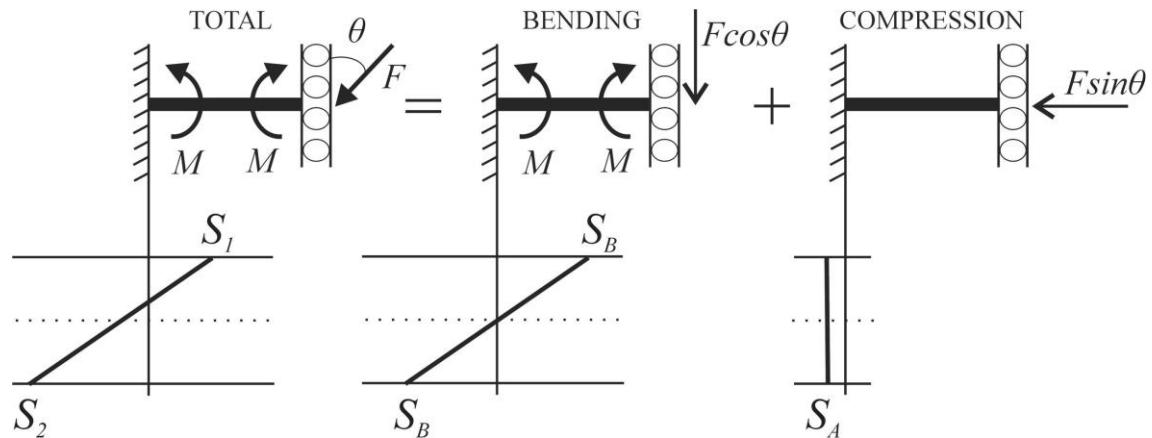
Maximum tensile stress in the struts:

$$S_1 = S_B - S_A = F \left(\frac{16L \cos \theta}{\pi d^3} - \frac{4 \sin \theta}{\pi d^2} \right)$$



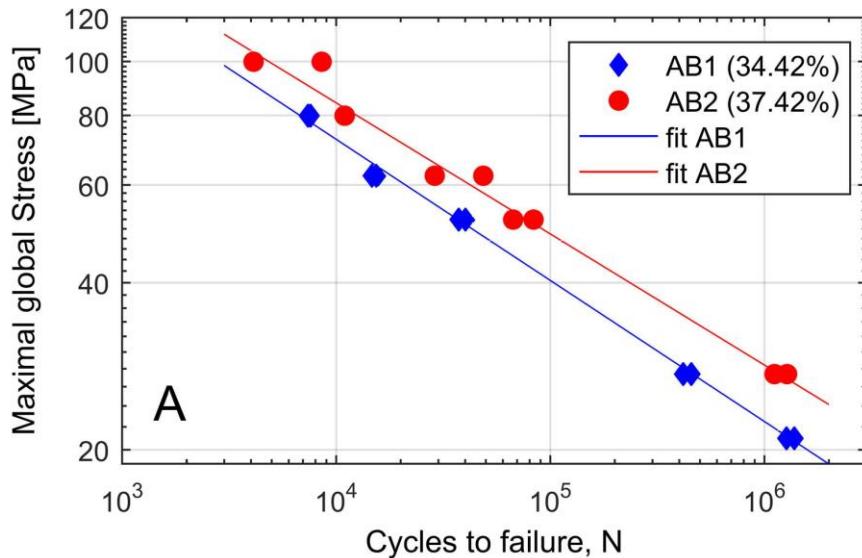
$$F = \frac{F_{tot}}{\text{number of load bearing struts}}$$

Compression-compression formulation

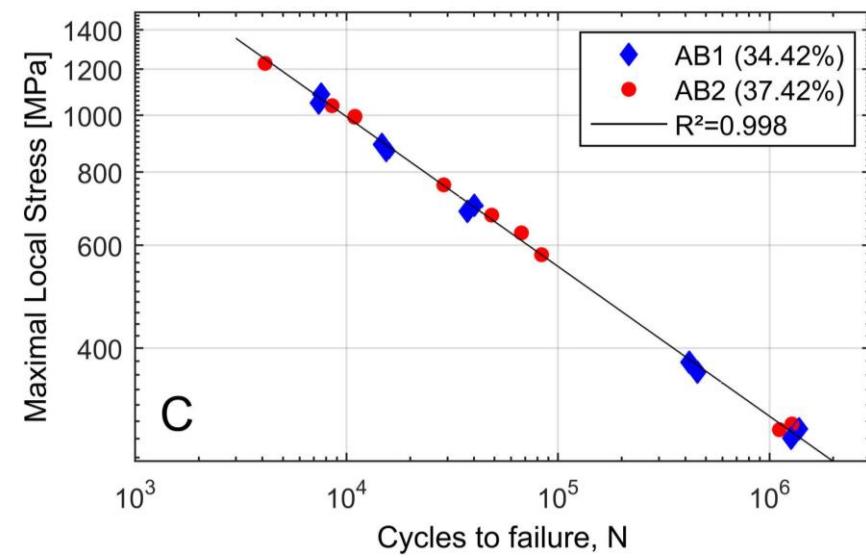


Fatigue performance: LSM vs global

Ti-6Al-4V

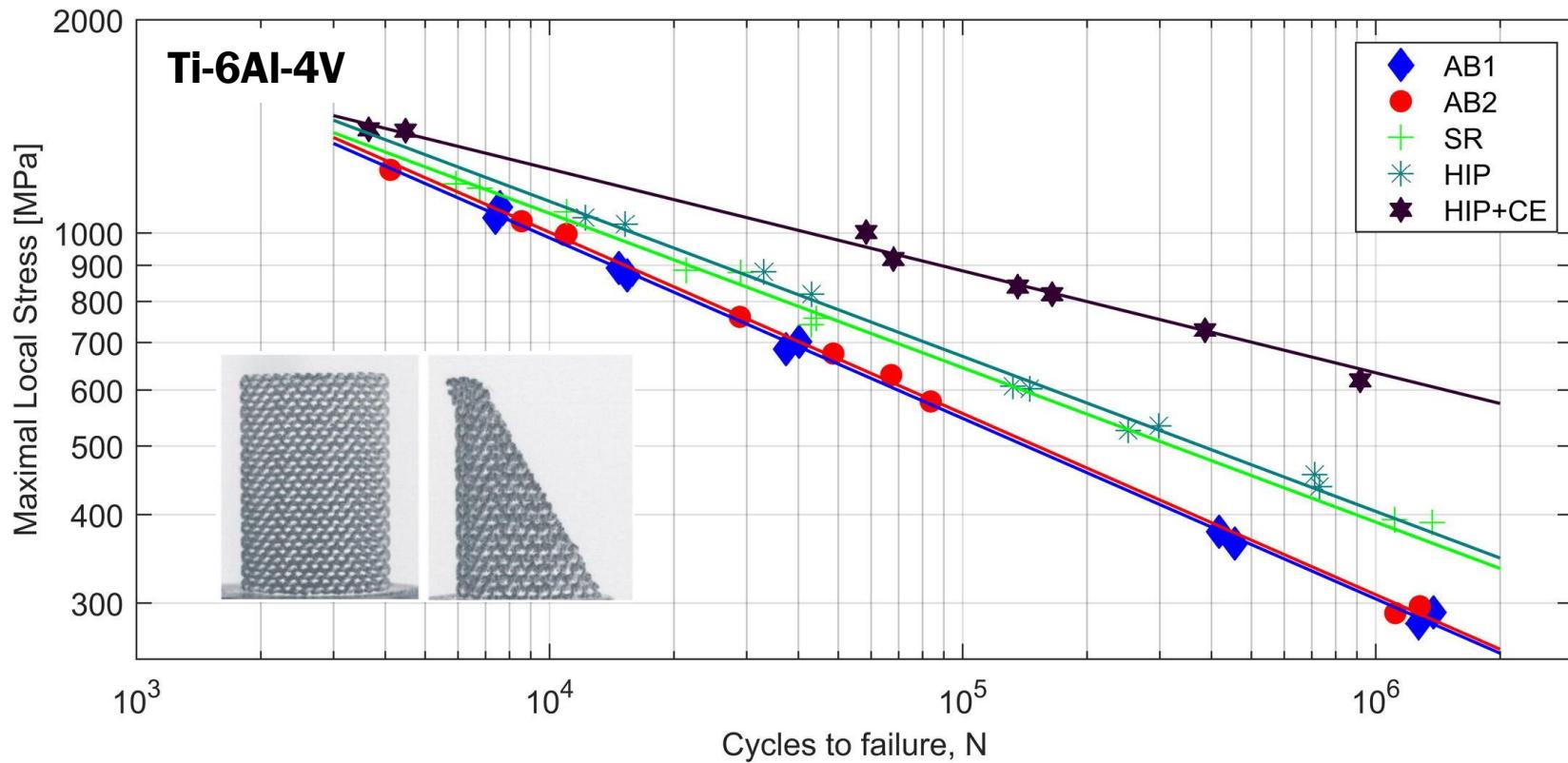


Global method



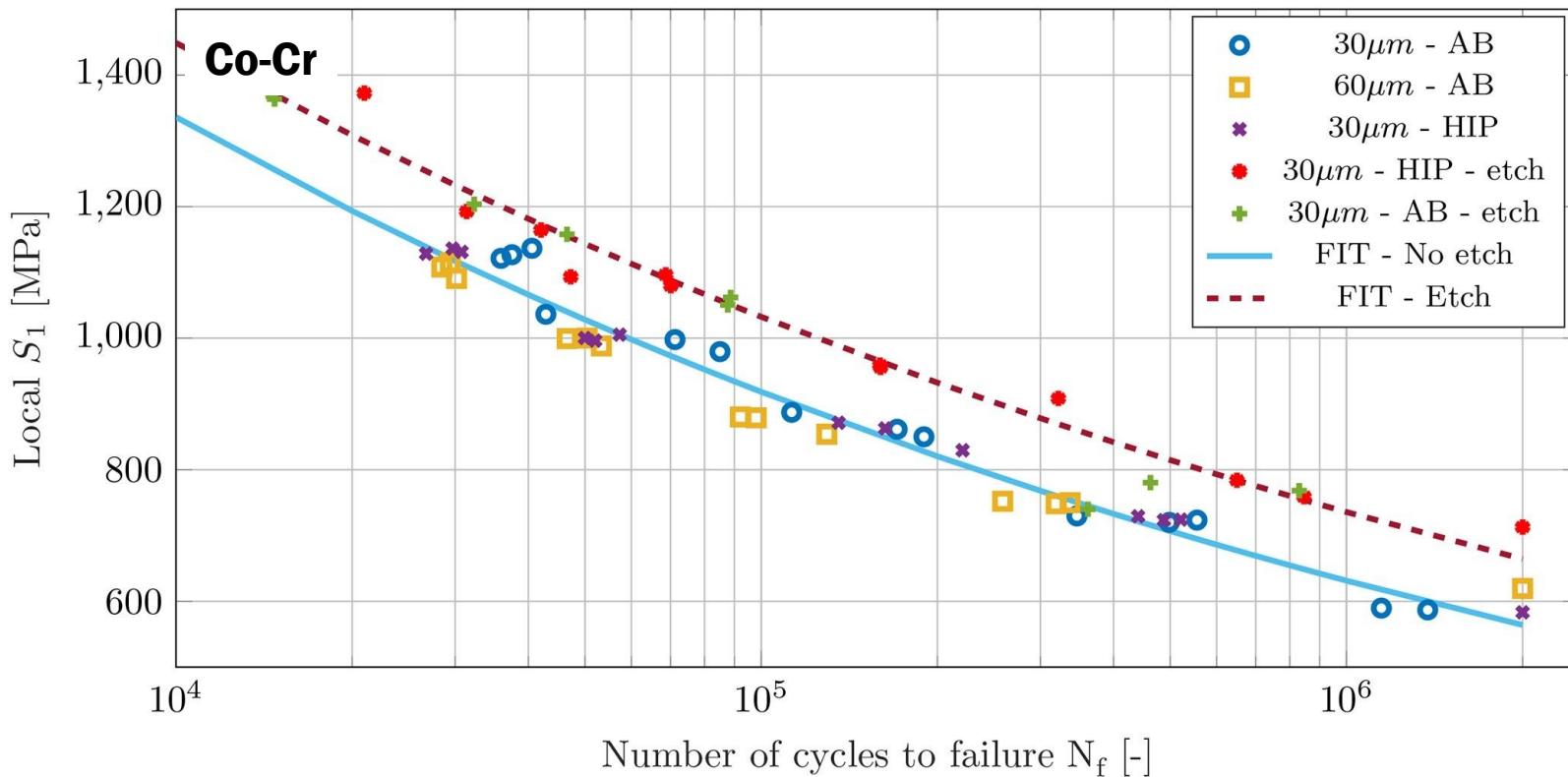
Local Stress Method

Fatigue performance: effect of post-treatments



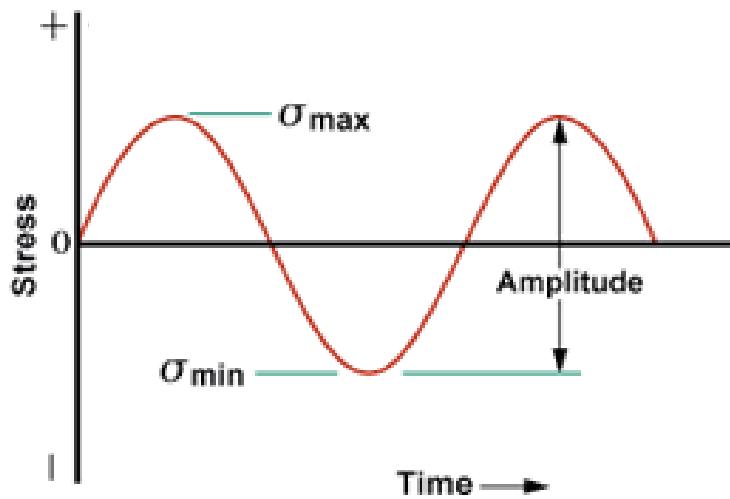
Chemical etching significantly improves the mechanical efficiency of the structure. In Ti64, SR and HIP improve the fatigue life to some extent.

Fatigue performance: effect of post-treatments

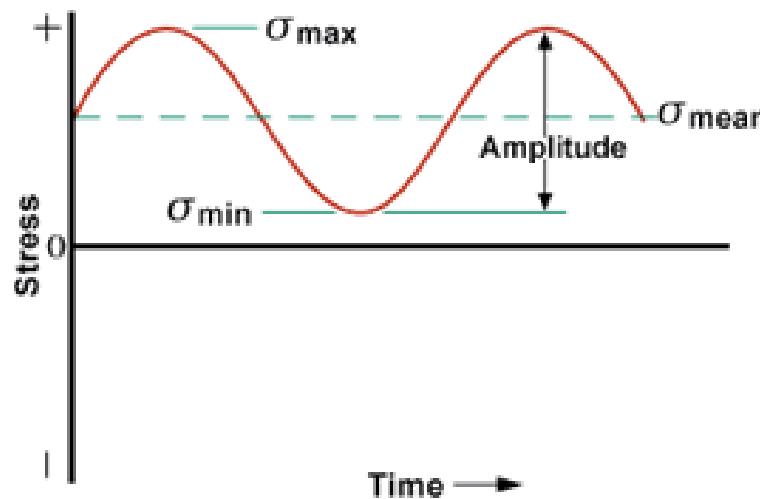


Chemical etching significantly improves the mechanical efficiency of the structure. In Co-Cr, there is no clear effect of HIP on the fatigue life.

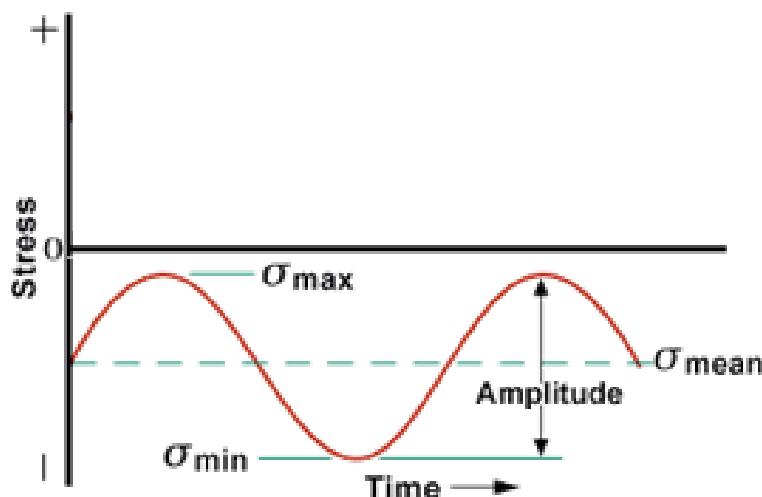
Fatigue performance: effect of loading conditions



Tension-compression



Tension-tension



Compression-compression

Local stress method - compression

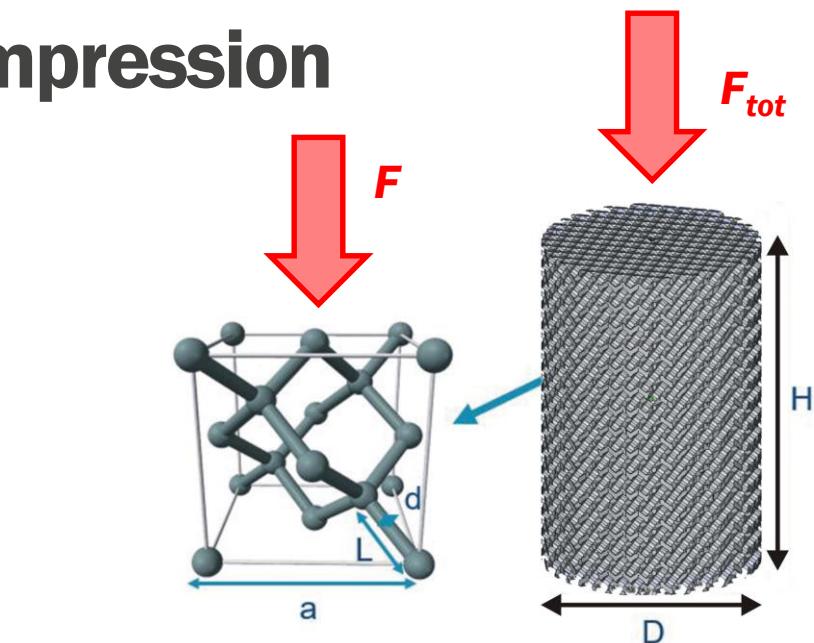
$$S_B = \frac{My}{I} = \frac{FLd \cos \theta}{4I} \quad \text{with:}$$

$$I = \frac{\pi d^4}{64} \quad y = \frac{d}{2} \quad M = \frac{1}{2} FL \cos \theta$$

$$S_A = \frac{F \sin \theta}{A} \quad \text{with} \quad A = \frac{\pi d^2}{4}$$

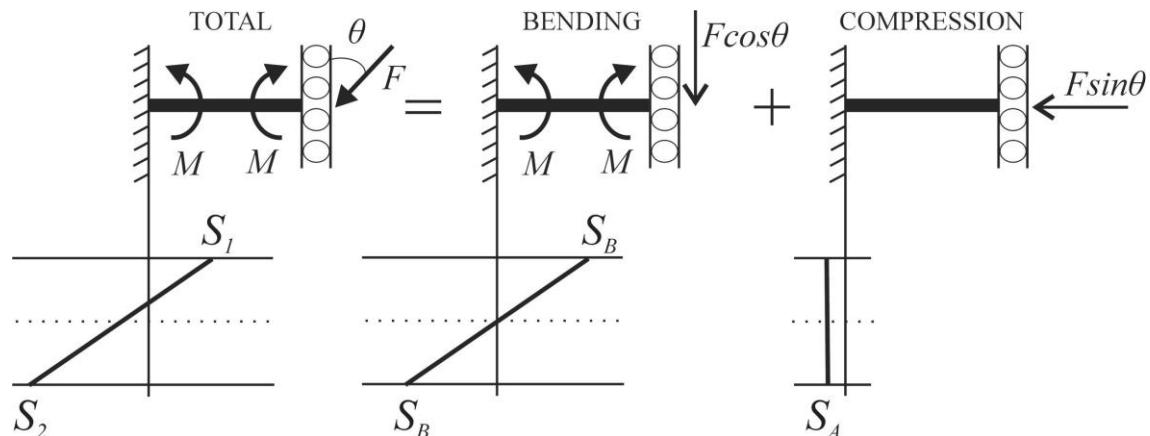
Maximum tensile stress in the struts:

$$S_1 = S_B - S_A = F \left(\frac{16L \cos \theta}{\pi d^3} - \frac{4 \sin \theta}{\pi d^2} \right)$$



$$F = \frac{F_{tot}}{\text{number of load bearing struts}}$$

Compression-compression
formulation



Local stress method - tension

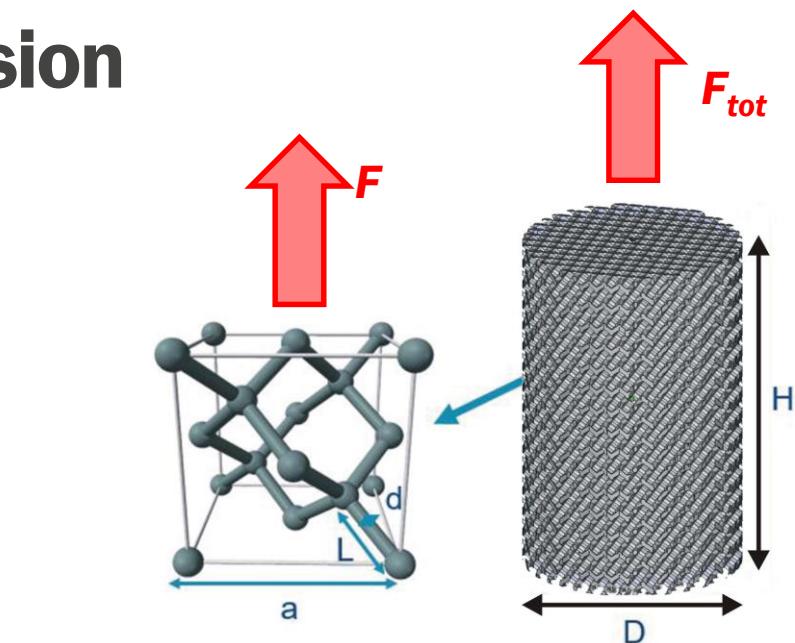
$$S_B = \frac{My}{I} = \frac{FLd \cos \theta}{4I} \quad \text{with:}$$

$$I = \frac{\pi d^4}{64} \quad y = \frac{d}{2} \quad M = \frac{1}{2} FL \cos \theta$$

$$S_A = \frac{F \sin \theta}{A} \quad \text{with} \quad A = \frac{\pi d^2}{4}$$

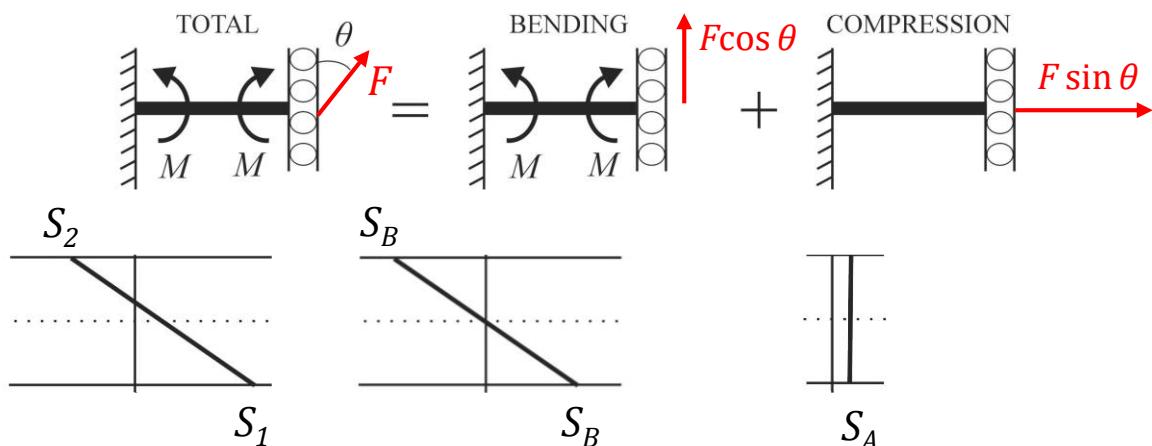
Maximum tensile stress in the struts:

$$S_1 = S_B + S_A = F \left(\frac{16L \cos \theta}{\pi d^3} + \frac{4 \sin \theta}{\pi d^2} \right)$$

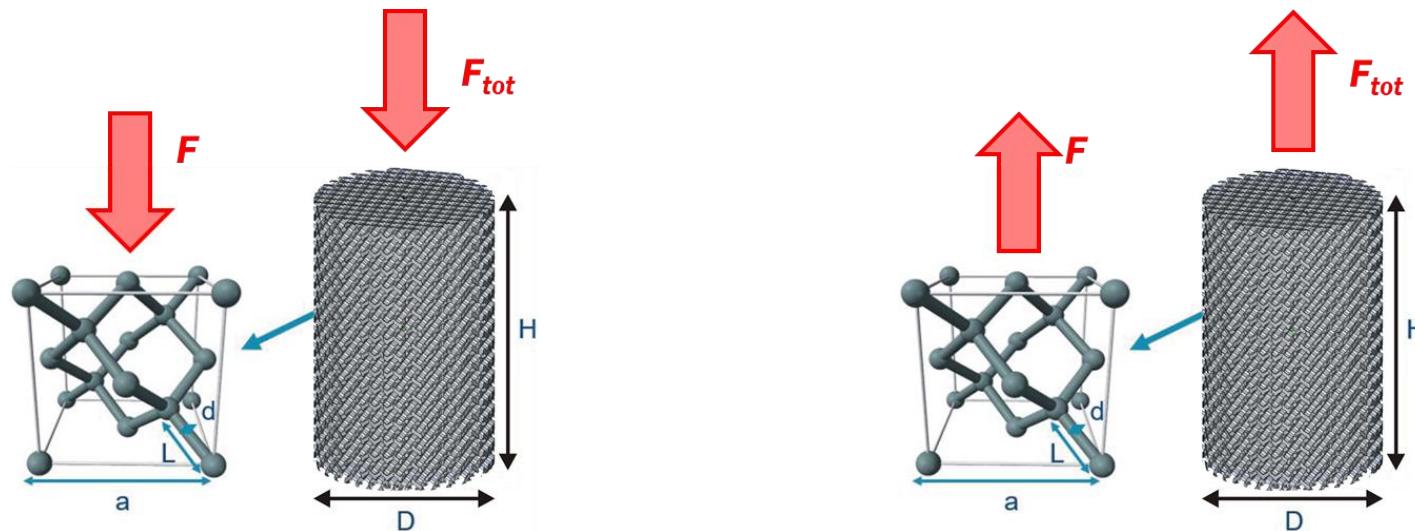


$$F = \frac{F_{tot}}{\text{number of load bearing struts}}$$

Tension-tension formulation



Fatigue performance: effect of loading conditions

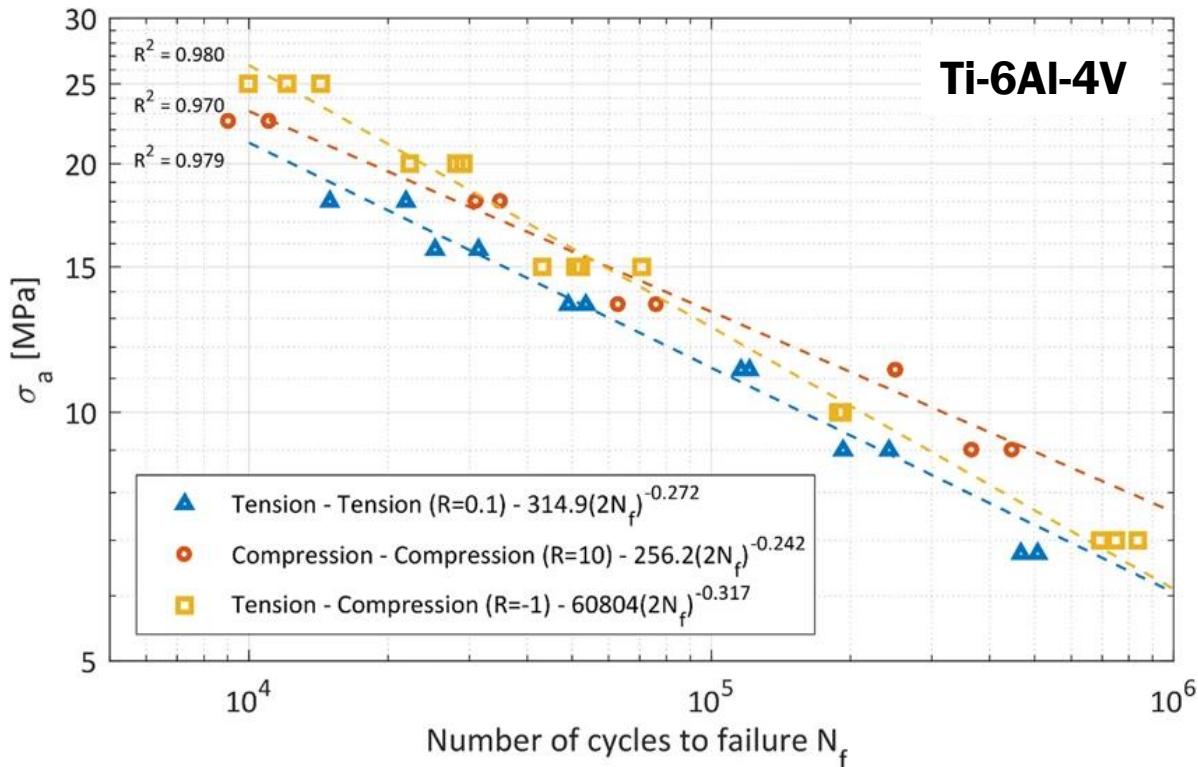


$$S_1 = S_B - S_A = F \left(\frac{16L \cos \theta}{\pi d^3} - \frac{4 \sin \theta}{\pi d^2} \right) < S_1 = S_B + S_A = F \left(\frac{16L \cos \theta}{\pi d^3} + \frac{4 \sin \theta}{\pi d^2} \right)$$

The same absolute global stress value generates a higher maximum tensile stress in tension than in compression.

⇒ **Lattice structures have a shorter fatigue life when they are loaded in tension than when they are loaded in compression**

Fatigue performance: effect of loading conditions



The same absolute global stress value generates a higher maximum tensile stress in tension than in compression.

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